

Symmetry of the Linearized Boltzmann Equation and Its Application

Shigeru Takata

Received: 18 December 2008 / Accepted: 17 July 2009 / Published online: 1 August 2009
© Springer Science+Business Media, LLC 2009

Abstract A symmetric relation of macroscopic quantities between two different steady problems of the linearized Boltzmann equation is derived. A few applications to half-space problems are presented first. Then, for the gas in bounded or unbounded domains such that solid bodies or condensed phases are confined in a finite region, general representations of the mass, momentum, and heat fluxes through the boundary (possibly at a point on or on a part of it) are derived from the symmetric relation linked to the separability of boundary data. This result implies a reduction of the original problem to a single elemental problem in the same domain, as far as the fluxes are concerned. Many applications are also presented.

Keywords Boltzmann equation · Reciprocity · Green function · Knudsen layer

1 Introduction

The linearized Boltzmann equation is widely used for the study of a slow rarefied gas flow or a gas in a micro scale system such as a micro channel, an aerosol particle, etc. One of the interesting features in such gas systems is that fluid-dynamical and thermal phenomena are mutually inductive. For instance, a net heat flow through a channel is induced in the Poiseuille flow, while a net mass flow is induced by a temperature gradient along the wall (the thermal transpiration [1]). Those phenomena are sometimes related to each other, e.g., the former net flow is identical to the latter up to the dimensional factor in the above example [2]. The thermal polarization [3, 4] and thermophoresis [5–7] of a particle are other examples of the mutually inductive phenomena expected to be related to each other [8, 9]. In the present paper, we investigate what kind of relation holds in general between two problems described by the steady linearized Boltzmann equation. We will eventually derive a useful representation of the fluxes through the boundary for quite general situations.

S. Takata (✉)

Department of Mechanical Engineering and Science, (also Advanced Research Institute of Fluid Engineering and Science), Graduate School of Engineering, Kyoto University, Kyoto 606-8501, Japan
e-mail: takata@aero.mbox.media.kyoto-u.ac.jp

First in Sect. 2, we consider two time-independent boundary-value problems of the linearized Boltzmann equation in the same domain. The domain may be bounded or unbounded, and its boundary is composed of two parts: one representing a simple solid surface or an interface with the condensed phase of the gas (the *real boundary*, for short) and the other representing an artificial boundary set in the interior of the gas, which includes the specular and periodic boundaries as a special case, or a far field in an unbounded domain (the *imaginary boundary*, for short). We derive a *symmetric relation of averaged macroscopic quantities* between the two problems by integrating both in space and molecular velocity a product of one solution and the other with reversed molecular velocity. The relation is obtained by the self-adjointness and parity of collision operator with the aid of the condition of detailed balance on the real boundary and its extension assumed on the artificial boundary, provided that the velocity distribution function approaches the given data sufficiently fast at a far distance for an unbounded domain. A concrete set of general situations in which the relation holds is also presented. These situations will be dealt with in the rest of the paper.

One of the situations where the symmetric relation holds is a spatially one-dimensional half-space problem of the linearized Boltzmann equation. In Sect. 3, we present some applications to half-space problems occurring in the study of the Knudsen layer structure [10–12]. In Sects. 3.1 and 3.2 we show the recovery of known relations for the so-called slip and jump coefficients, while in Sect. 3.3 we provide unknown relations for jump coefficients and a couple of new numerical data by use of them. In the half-space problems, the boundary data on the real boundary affect the state of the gas in a far field. On the other hand, for bounded or unbounded domains such that the real boundary is confined in a finite region, we may deal with the boundary data separately, place by place, and consider a response of the system to elemental sources put on the boundary. We shall call the response the *Green function* in the present paper. In such domains, we can assure the symmetric relation to hold by a rather simple argument based on the Stokes set of equations. In Sect. 4, we present a fruitful consequence from the separability of the boundary data linked to the symmetric relation. To be specific, after a preliminary argument in Sect. 4.2, we present general expressions of the mass, momentum, and heat fluxes on the boundary in Sect. 4.3, which we call the representation theorem and is the second main outcome of the present paper. The obtained representation is not a mere superposition of the Green function. It tells that *the problem of finding a flux through the boundary (possibly a part of or even a point of it) is reduced to finding the Green function for the elemental source corresponding to that flux put on the interested boundary*. The elemental sources are a collision invariant distributed uniformly or linearly in space on the boundary (possibly on a part of or even at a point on it). Their correspondence to the fluxes will be clear in the course of discussion in Sects. 4.2 and 4.3. We also show corollaries of the theorem on the reciprocity of the fluxes induced by the Green functions in Sect. 4.4. This property will become important, especially when linked to the entropy theory to be developed in a separate paper. Various applications of the theorem will be shown in Sect. 5.

Our Green function is not the ones discussed in [13, 14]. The source is not in the equation but in the boundary condition as the macroscopically meaningful elemental data. This is intended to reflect the fact that the boundary data are given in accordance with the state of the surroundings at the macroscopic level. Our purpose is not to represent the solution itself as a superposition of the Green functions but to show the reduction of the original problem to obtaining a single elemental solution corresponding to the interested fluxes through the boundary. Simple analogues might come to mind. For instance, the Green reciprocity theorem [15–17] for the Laplace equation is known in the electrostatics for conductors and in the theory of heat conduction. For the continuum or near continuum gas (i.e., the system with

small Knudsen numbers) in the linear regime, the flow field is described by the Stokes equation for the incompressible fluid and the temperature field by the Laplace equation [10–12]. From this point of view, the representation theorem may be regarded as the extension of the classical reciprocity to arbitrary Knudsen numbers, though it includes more. We will give a comment on this aspect in Corollary 9 in Sect. 5. We also mention a simple analogue in the theory of radiative transfer [18] to the example of a half-space problem in Sect. 3.2.

2 Symmetry of the Steady Linearized Boltzmann Equation

2.1 Problem

We start with a physically rather abstract or seemingly artificial formulation of the problem.

Let us denote by \mathbf{x} and $\boldsymbol{\zeta}$ the dimensionless position and molecular velocity. We shall consider a function $\phi(\mathbf{x}, \boldsymbol{\zeta})$ in a certain domain of \mathbf{x} , say D , and for the whole space of $\boldsymbol{\zeta}$ that satisfies the steady, inhomogeneous linearized Boltzmann equation:

$$\zeta_i \frac{\partial \phi}{\partial x_i} + F_i \frac{\partial \phi}{\partial \zeta_i} = \frac{1}{K} \mathcal{L}(\phi) + I. \tag{1}$$

Here \mathbf{F} is a given vector depending on \mathbf{x} and $\boldsymbol{\zeta}$ and satisfies the condition $\frac{\partial}{\partial \zeta_i} F_i E = 0$ with $E(\boldsymbol{\zeta}) = \pi^{-3/2} \exp(-|\boldsymbol{\zeta}|^2)$, K is an arbitrary positive constant ($0 < K < \infty$), and I is a given function of \mathbf{x} and $\boldsymbol{\zeta}$. The explicit form of the linearized collision integral $\mathcal{L}(\phi)$ is suppressed, because \mathcal{L} is required only to have the following properties in the discussions:

- (i) $\mathcal{L}(\cdot)$ commutes with the parity operator acting on $\boldsymbol{\zeta}$:

$$\mathcal{L}(\Phi)^- = \mathcal{L}(\Phi^-) \quad \text{for any } \Phi, \tag{2a}$$

where the function with superscript $-$ is defined as $\Psi^-(\mathbf{x}, \boldsymbol{\zeta}) \equiv \Psi(\mathbf{x}, -\boldsymbol{\zeta})$.

- (ii) $\mathcal{L}(\cdot)$ is self-adjoint:

$$\langle \Phi \mathcal{L}(\Psi) \rangle = \langle \Psi \mathcal{L}(\Phi) \rangle \quad \text{for any } \Phi \text{ and } \Psi, \tag{2b}$$

where the brackets $\langle \cdot \rangle$ indicate the following moment with respect to $\boldsymbol{\zeta}$:

$$\langle \Phi \rangle = \int \Phi(\boldsymbol{\zeta}) E(\boldsymbol{\zeta}) d\boldsymbol{\zeta}.$$

- (iii) $\mathcal{L}(\Phi) = 0$ holds if and only if Φ is a linear combination of 1, $\boldsymbol{\zeta}$, and $|\boldsymbol{\zeta}|^2$.

- (iv) $\mathcal{L}(\cdot)$ is non-positive:

$$\langle \Phi \mathcal{L}(\Phi) \rangle \leq 0 \quad \text{for any } \Phi, \tag{2c}$$

and the equality holds if and only if Φ is a linear combination of 1, $\boldsymbol{\zeta}$, and $|\boldsymbol{\zeta}|^2$.

The following obvious properties will be frequently used in the sequel:

$$(\Phi^-)^- = \Phi, \quad \langle \Phi \rangle = \langle \Phi^- \rangle \quad \text{for any } \Phi. \tag{3}$$

On a part of the boundary ∂D , which we denote by ∂D_w , ϕ obeys the following condition:

$$\phi = g_w + \int_{\zeta_n^* < 0} \frac{|\zeta_n^*| E(\boldsymbol{\zeta}^*)}{|\zeta_n| E(\boldsymbol{\zeta})} R(\boldsymbol{\zeta}^*, \boldsymbol{\zeta}; \mathbf{x}) (\phi^* - g_w^*) d\boldsymbol{\zeta}^*, \quad \zeta_n > 0, \tag{4}$$

where $\zeta_n = \boldsymbol{\zeta} \cdot \mathbf{n}$ with \mathbf{n} being the unit vector inward normal to the boundary ∂D_w at position \mathbf{x} , g_w is a given function of \mathbf{x} and $\boldsymbol{\zeta}$ to be specified soon later in (5), and ϕ^* and g_w^* stand for $\phi(\mathbf{x}, \boldsymbol{\zeta}^*)$ and $g_w(\mathbf{x}, \boldsymbol{\zeta}^*)$ respectively. The kernel R is a given function having the properties summarized in Appendix A, where and in what follows we denote by R_{CR} the kernel satisfying the condition $\langle \zeta_n \phi \rangle = 0$ and by R_{PR} the kernel free from this condition. The required properties are different between R_{CR} and R_{PR} . The function $g_w(\mathbf{x}, \boldsymbol{\zeta})$ is defined by

$$g_w = \begin{cases} 2\boldsymbol{\zeta} \cdot \mathbf{u}_w + (|\boldsymbol{\zeta}|^2 - \frac{5}{2})\tau_w & \text{if } R = R_{CR}, \\ P_w + 2\boldsymbol{\zeta} \cdot \mathbf{u}_w + (|\boldsymbol{\zeta}|^2 - \frac{5}{2})\tau_w & \text{if } R = R_{PR}, \end{cases} \tag{5}$$

where \mathbf{u}_w , τ_w , and P_w are given constants in $\boldsymbol{\zeta}$ with $\mathbf{u}_w \cdot \mathbf{n} = 0$. In general, \mathbf{u}_w , τ_w , and P_w depend on the position \mathbf{x} on ∂D_w . Note that g_w is defined for the whole range of $\boldsymbol{\zeta}$.

Physically, ∂D_w is the part corresponding to the *real boundary*, which is the surface of a simple solid body (a simple surface, for short) or the interface with the condensed phase, i.e., liquid or solid, of the gas (the interface, for short). On the former, there occurs no mass flow across the boundary (i.e., $\langle \zeta_n \phi \rangle = 0$), while on the latter it may occur in general. R_{CR} represents the reflection kernel for the former and R_{PR} that for the latter in the reference equilibrium state at rest. The deviation of the state of the boundary are represented by P_w , \mathbf{u}_w , and τ_w involved in g_w . The dependence of R on \mathbf{x} comes from the possible change of the boundary material and typically appears as the change of the accommodation coefficients.

From now on, we consider two functions ϕ^A and ϕ^B such that:

1. ϕ^A satisfies (1) and (4) with $F_i = \mathcal{F}_i$, $I = I^A$, and $g_w = g_w^A$,
2. ϕ^B satisfies (1) and (4) with $F_i = \mathcal{F}_i^-$, $I = I^B$, and $g_w = g_w^B$,

where R in (4) is common to ϕ^A and ϕ^B . The aim of Sect. 2 is to derive a symmetric identity between the global quantities of ϕ^A and of ϕ^B . For the moment, we proceed without any information about ϕ 's on the remaining part of the boundary ∂D , which we denote by ∂D_g , i.e., $\partial D_g = \partial D \setminus \partial D_w$.

2.2 Symmetric Relation

We first show a symmetric identity that can be obtained without specific information on ∂D_g . The presented form is a slight extension of those in the literature (e.g., [13, 19, 20]) mainly in the sense that I is arbitrary as far as the solution exists.

Proposition 1 *Consider the functions ϕ^A and ϕ^B such that:*

- (i) ϕ^A satisfies (1) and (4) with $F_i = \mathcal{F}_i$, $I = I^A$, and $g_w = g_w^A$,
- (ii) ϕ^B satisfies (1) and (4) with $F_i = \mathcal{F}_i^-$, $I = I^B$, and $g_w = g_w^B$,

where R in (4) is common to ϕ^A and ϕ^B . If the kernel R satisfies the condition of detailed balance [13, 21] (see Appendix B):

$$|\zeta_n^*| R(\boldsymbol{\zeta}^*, \boldsymbol{\zeta}; \mathbf{x}) E(\boldsymbol{\zeta}^*) = |\zeta_n| R(-\boldsymbol{\zeta}, -\boldsymbol{\zeta}^*; \mathbf{x}) E(\boldsymbol{\zeta}) \quad \text{for } \zeta_n > 0, \zeta_n^* < 0, \tag{6}$$

the following symmetric identity holds:

$$\begin{aligned} & \int_{\partial D_w} \langle \zeta_n g_w^{B-} \phi^A \rangle dS + \frac{1}{2} \int_{\partial D_g} \langle \zeta_n \phi^{B-} \phi^A \rangle dS - \int_D \langle I^{B-} \phi^A \rangle dx \\ &= \int_{\partial D_w} \langle \zeta_n g_w^{A-} \phi^B \rangle dS + \frac{1}{2} \int_{\partial D_g} \langle \zeta_n \phi^{A-} \phi^B \rangle dS - \int_D \langle I^{A-} \phi^B \rangle dx, \end{aligned} \tag{7}$$

where dS is the surface element at position \mathbf{x} .

Proof We first integrate (1) for ϕ^A (or ϕ^B) multiplied by $\phi^{B-}E$ (or $\phi^{A-}E$) over the whole space of ζ :

$$\left\langle \zeta_i \frac{\partial \phi^A}{\partial x_i} \phi^{B-} \right\rangle + \left\langle \mathcal{F}_i \frac{\partial \phi^A}{\partial \zeta_i} \phi^{B-} \right\rangle = \frac{1}{K} \langle \mathcal{L}(\phi^A) \phi^{B-} \rangle + \langle I^A \phi^{B-} \rangle, \tag{8a}$$

$$\left\langle \zeta_i \frac{\partial \phi^B}{\partial x_i} \phi^{A-} \right\rangle + \left\langle \mathcal{F}_i \frac{\partial \phi^B}{\partial \zeta_i} \phi^{A-} \right\rangle = \frac{1}{K} \langle \mathcal{L}(\phi^B) \phi^{A-} \rangle + \langle I^B \phi^{A-} \rangle. \tag{8b}$$

Subtracting (8b) from (8a) leads to, with the aid of (2a), (2b), (3), and the condition for F immediately after (1),

$$\frac{1}{2} \frac{\partial}{\partial x_i} \langle \zeta_i \phi^{B-} \phi^A \rangle + \langle I^{B-} \phi^A \rangle = \frac{1}{2} \frac{\partial}{\partial x_i} \langle \zeta_i \phi^{A-} \phi^B \rangle + \langle I^{A-} \phi^B \rangle. \tag{9}$$

Integrating (9) over the domain D results in the well-known identity (e.g., see [13]):

$$-\frac{1}{2} \int_{\partial D} \langle \zeta_n \phi^{B-} \phi^A \rangle dS + \int_D \langle I^{B-} \phi^A \rangle dS = -\frac{1}{2} \int_{\partial D} \langle \zeta_n \phi^{A-} \phi^B \rangle dS + \int_D \langle I^{A-} \phi^B \rangle dS. \tag{10}$$

Next, we split the surface integrals into those on ∂D_w and ∂D_g and transform the integral on the former. Simple rearrangement yields on ∂D_w

$$\langle \zeta_n \phi^{B-} \phi^A \rangle = \langle \zeta_n (\phi^{B-} - g_w^{B-}) (\phi^A - g_w^A) \rangle + \langle \zeta_n g_w^{B-} \phi^A \rangle + \langle \zeta_n \phi^{B-} g_w^A \rangle - \langle \zeta_n g_w^{B-} g_w^A \rangle.$$

The last term on the right-hand side vanishes because of (5) and $u_w \cdot n = 0$. Further, for R satisfying the condition of detailed balance, the first term on the right-hand side vanishes, as shown in Appendix B. Thus, the integrand of the surface integral on ∂D_w is reduced to

$$\langle \zeta_n \phi^{B-} \phi^A \rangle = \langle \zeta_n g_w^{B-} \phi^A \rangle - \langle \zeta_n g_w^{A-} \phi^B \rangle, \quad \langle \zeta_n \phi^{A-} \phi^B \rangle = \langle \zeta_n g_w^{A-} \phi^B \rangle - \langle \zeta_n g_w^{B-} \phi^A \rangle.$$

Substitution into (10) yields the desired identity. □

2.2.1 Condition on ∂D_g

We now introduce the condition for ϕ on ∂D_g to refine Proposition 1, and thus ϕ ought to be considered as a solution of the boundary-value problem of (1), (4), and the condition (11) on ∂D_g below. We shall consider three types of conditions on ∂D_g and correspondingly split it into three parts: $\partial D_g^{(i)}$, $\partial D_g^{(ii)}$, and $\partial D_g^{(iii)}$. Each part is defined as follows:

- (i) $\partial D_g^{(i)}$ is the part where ϕ for the inward direction of ζ is given:¹

$$\phi(x, \zeta) = h_{in}(x, \zeta) \quad \text{for } \zeta_n > 0, x \in \partial D_g^{(i)}. \tag{11a}$$

Here h_{in} is a given function defined for $\zeta_n > 0$. For later discussions, we extend the range of this function to the whole space of ζ and denote the extended function by $h(x, \zeta)$. The way of extension is arbitrary.

¹Mathematically, the conditions (4) and (11a) do not exclude each other, so that there may be a part of the boundary which can be regarded as in ∂D_w or in $\partial D_g^{(i)}$. However, there arises no difference to be cared.

(ii) $\partial D_g^{(ii)}$ is the part that reaches infinity when D is an unbounded domain. The asymptotic form of ϕ for large $|\mathbf{x}|$ is given:

$$\phi(\mathbf{x}, \boldsymbol{\zeta}) \rightarrow h(\mathbf{x}, \boldsymbol{\zeta}) \quad \text{as } |\mathbf{x}| \rightarrow \infty. \tag{11b}$$

Note that $h(\mathbf{x}, \boldsymbol{\zeta})$ is defined for the whole range of $\boldsymbol{\zeta}$ and satisfies (1) by definition.

(iii) $\partial D_g^{(iii)}$ is the part where ϕ for the inward direction of $\boldsymbol{\zeta}$ at position \mathbf{x} is related to that for the outward direction of $\boldsymbol{\zeta}'$ at position \mathbf{x}' :

$$\phi(\mathbf{x}, \boldsymbol{\zeta}) = h(\mathbf{x}, \boldsymbol{\zeta}) + \int_{\partial D_g^{(iii)}} \int_{\zeta'_n < 0} P(\mathbf{x}', \boldsymbol{\zeta}', \mathbf{x}, \boldsymbol{\zeta})(\phi' - h') d\boldsymbol{\zeta}' dS' \quad \text{for } \zeta_n > 0, \mathbf{x} \in \partial D_g^{(iii)}, \tag{11c}$$

where $h(\mathbf{x}, \boldsymbol{\zeta})$ is a given function defined for the whole range of $\boldsymbol{\zeta}$ and for any $\mathbf{x} \in \partial D_g^{(iii)}$, $\zeta'_n = \boldsymbol{\zeta}' \cdot \mathbf{n}'$, and \mathbf{n}' and dS' are the inward unit vector normal to $\partial D_g^{(iii)}$ and the surface element at point \mathbf{x}' respectively. ϕ' and h' stand for $\phi(\mathbf{x}', \boldsymbol{\zeta}')$ and $h(\mathbf{x}', \boldsymbol{\zeta}')$ respectively. P is a given function defined for $\zeta_n > 0$ and $\zeta'_n < 0$ that prescribes the relation between two points \mathbf{x} and \mathbf{x}' on $\partial D_g^{(iii)}$. Here we restrict P to the functions that meet the following conditions (an extension of the conditions in Appendices A and B):

(a) For $\zeta_n > 0$ and $\zeta'_n < 0$,

$$|\zeta_n| E(\boldsymbol{\zeta}) P(\mathbf{x}', \boldsymbol{\zeta}', \mathbf{x}, \boldsymbol{\zeta}) = |\zeta'_n| E(\boldsymbol{\zeta}') P(\mathbf{x}, -\boldsymbol{\zeta}, \mathbf{x}', -\boldsymbol{\zeta}').$$

(b) For a certain given function $g_0(\mathbf{x}, \boldsymbol{\zeta}) \geq 0$ defined for $\zeta_n > 0$ and $\mathbf{x} \in \partial D_g^{(iii)}$, the following relation holds:

$$1 = g_0(\mathbf{x}, \boldsymbol{\zeta}) + \int_{\partial D_g^{(iii)}} \int_{\zeta'_n < 0} P(\mathbf{x}', \boldsymbol{\zeta}', \mathbf{x}, \boldsymbol{\zeta}) d\boldsymbol{\zeta}' dS' \quad \text{for } \zeta_n > 0.$$

(c) $P(\mathbf{x}', \boldsymbol{\zeta}', \mathbf{x}, \boldsymbol{\zeta}) \geq 0$, and it is not identically zero.

The property (a) is an extension of the condition of detailed balance to among different points on the boundary. It should be noted that the uniqueness condition corresponding to the third property of R_{CR} and R_{PR} in Appendix A is not required of P . Thus, the specular and periodic type boundaries are classified into $\partial D_g^{(iii)}$. Due to the property (c), (11a) is excluded from (11c).

Some physical comments would be in order on the boundary conditions. As mentioned before, ∂D_w is the part corresponding to the *real boundary*, i.e., a simple surface or the interface. On the former, there occurs no mass flow across the boundary, while on the latter it may occur in general. The kernel R_{CR} represents the reflection rule on the former and R_{PR} on the latter. It is important to notice that the specular reflection condition is excluded from the part ∂D_w because of the third property of R_{CR} and R_{PR} in Appendix A. The remaining part ∂D_g is the *imaginary boundary*, which is set inside a gas. The specular reflection and periodic boundary conditions are a typical example of the part $\partial D_g^{(iii)}$. It is readily checked that they have the properties required of P . Arbitrariness of h on $\partial D_g^{(i)}$ and $\partial D_g^{(iii)}$ aims at the application to a rather artificial problem setting often adopted in numerical simulations.

We close the present subsection with the following lemma:

Lemma 1 Let ϕ^A and ϕ^B obey the condition (11) with $h = h^A$ and $h = h^B$ respectively, where P in (11c) is common to ϕ^A and ϕ^B . Then the following identity holds:

$$\int_{\partial D_g^{(i)} \cup \partial D_g^{(iii)}} \langle \zeta_n (\phi^{A-} - h^{A-}) (\phi^B - h^B) \rangle dS = 0, \quad (12)$$

where h^A and h^B on $\partial D_g^{(i)}$ are an arbitrary extension of h_{in}^A and h_{in}^B to the whole range of ζ .

Proof For brevity, we denote $\phi^\alpha - h^\alpha$ by ψ^α ($\alpha = A, B$). Obviously $\psi^\alpha = 0$ for $\zeta_n > 0$ on $\partial D_g^{(i)}$. The integrand on $\partial D_g^{(i)}$ is readily seen to vanish by a change of variables:

$$\langle \zeta_n \psi^{A-} \psi^B \rangle = \int_{\zeta_n > 0} \zeta_n \psi^{A-} \psi^B E d\zeta - \int_{\zeta_n > 0} \zeta_n \psi^A \psi^{B-} E d\zeta = 0,$$

and (12) is reduced to the surface integral on $\partial D_g^{(iii)}$. For the reduced integral

$$\int_{\partial D_g^{(iii)}} \langle \zeta_n \psi^{A-} \psi^B \rangle dS = \int_{\partial D_g^{(iii)}} \int_{\zeta_n > 0} \zeta_n \psi^{A-} \psi^B E d\zeta dS - \int_{\partial D_g^{(iii)}} \int_{\zeta_n > 0} \zeta_n \psi^A \psi^{B-} E d\zeta dS,$$

we use (11c) for ψ^B in the first term and that for ψ^A in the second term on the right-hand side. Then, by the use of the property (a) of P , the first and second terms are seen to cancel out each other in a way similar to the proof of Lemma 4 in Appendix B. \square

2.2.2 Symmetric Relation

We now present the first main outcome of the present paper, which is the refinement of Proposition 1 due to Lemma 1:

Proposition 2 (Symmetric relation) Consider the solutions ϕ^A and ϕ^B of the boundary-value problem (1), (4), and (11) such that:

- (i) ϕ^A is a solution in the case of $I = I^A$, $g_w = g_w^A$, $F_i = \mathcal{F}_i$, and $h = h^A$,
- (ii) ϕ^B is a solution in the case of $I = I^B$, $g_w = g_w^B$, $F_i = \mathcal{F}_i^-$, and $h = h^B$,

where R in (4) and P in (11c) are common to the problems. Then:

1. If the kernel R on ∂D_w satisfies the condition of detailed balance (6), and
2. if the part $\partial D_g^{(ii)}$ is absent or ϕ^A and ϕ^B approach h^A and h^B sufficiently fast in (11b) so that

$$\int_{\partial D_g^{(ii)}} \langle \zeta_n (\phi^{A-} - h^{A-}) (\phi^B - h^B) \rangle dS = 0, \quad (13)$$

the following equality holds:

$$\begin{aligned} & \int_{\partial D_w} \langle \zeta_n g_w^{B-} \phi^A \rangle dS + \int_{\partial D_g} \langle \zeta_n h^{B-} \phi^A \rangle dS - \frac{1}{2} \int_{\partial D_g} \langle \zeta_n h^{B-} h^A \rangle dS - \int_D \langle I^{B-} \phi^A \rangle dx \\ & = \int_{\partial D_w} \langle \zeta_n g_w^{A-} \phi^B \rangle dS + \int_{\partial D_g} \langle \zeta_n h^{A-} \phi^B \rangle dS - \frac{1}{2} \int_{\partial D_g} \langle \zeta_n h^{A-} h^B \rangle dS - \int_D \langle I^{A-} \phi^B \rangle dx. \end{aligned} \quad (14)$$

It should be reminded that h^A and h^B on $\partial D_g^{(i)}$ may be any extension of h_{in}^A and h_{in}^B to the whole range of ζ [see (i) in Sect. 2.2.1].²

Proof The refinement is achieved by the reduction of $\langle \zeta_n \phi^{A-} \phi^B \rangle$ on ∂D_g due to Lemma 1:

$$\begin{aligned} \int_{\partial D_g} \langle \zeta_n \phi^{A-} \phi^B \rangle dS &= \int_{\partial D_g} \left(\langle \zeta_n h^{A-} \phi^B \rangle + \langle \zeta_n \phi^{A-} h^B \rangle - \langle \zeta_n h^{A-} h^B \rangle \right) dS \\ &\quad + \int_{\partial D_g^{(ii)}} \langle \zeta_n (\phi^{A-} - h^{A-})(\phi^B - h^B) \rangle dS. \end{aligned} \quad \square$$

It should be noted that l and h are not restricted to any specific form as far as the solution ϕ exists (see the example in Sect. 3.3).

The condition (13) is fulfilled for bounded domains. It is also fulfilled for unbounded domains at least in the following two general situations for $F = 0$:

1. D is a spatially one-dimensional half-space.
2. D is an unbounded three dimensional domain and ∂D_w is confined in a finite region, so that $\partial D_g = \partial D_g^{(ii)}$.

The first case is obvious because the area of surface integral does not change in passing to the limit $|x| \rightarrow \infty$, and ϕ^A and ϕ^B in the surface integral on ∂D_g may be replaced by h^A and h^B . The second case is not trivial and is due to Lemma 2 to be shown soon later.³

In the rest of the paper, we work on bounded domains and unbounded domains in the situations raised above. We first present some applications to the first case, i.e., one-dimensional half-space problems, in Sect. 3. The first case is not merely simple but also requires a separate discussion from the second case. It is due to the fact that the source on the real boundary g_w does affect the state of the gas in a far field in that case and thus h cannot be given independent of g_w [22]. In contrast, we may separately deal with the sources on the boundaries in the second case, i.e., the three dimensional unbounded domain, and consequently can establish a unified theoretical framework with the case of bounded domain. We present this unified approach in Sect. 4, which leads to the second main outcome of the present paper on general representations of the mass, momentum, and heat fluxes passing through the boundary. We will show its applications in Sect. 5.

We close the present subsection with the announced lemma:

Lemma 2 *Let ϕ^A and ϕ^B satisfy (1) with $F = 0$ in an unbounded domain D with $\partial D_g = \partial D_g^{(ii)}$, where $h = h^A$ and h^B respectively in (11b). Then the following equality holds:*

$$\int_{\partial D_g} \langle \zeta_n (\phi^{A-} - h^{A-})(\phi^B - h^B) \rangle dS = 0.$$

²The respective values of the second and third terms of each side of (14) depend on the way of extension of h_{in}^A and h_{in}^B . However, the following relation always holds:

$$\int_{\partial D_g} \langle \zeta_n h^{B-} \phi^A \rangle dS - \frac{1}{2} \int_{\partial D_g} \langle \zeta_n h^{B-} h^A \rangle dS = \int_{\partial D_g} \langle \zeta_n h^{A-} \phi^B \rangle dS - \frac{1}{2} \int_{\partial D_g} \langle \zeta_n h^{A-} h^B \rangle dS.$$

³As is clear from the proof of Lemma 2, the key estimate (33) in [19] is incorrect.

Proof Let us denote $\phi^\alpha - h^\alpha$ by ψ^α ($\alpha = A, B$). Since, by definition, h^α is a solution of (1)⁴ with $\mathbf{F} = 0$, ψ^α is a solution of (1) with $\mathbf{F} = 0$ and $I^\alpha = 0$ and tends to vanish as $|\mathbf{x}| \rightarrow \infty$. Now consider a sufficiently large sphere that contains the real boundary ∂D_w inside. We denote by S_g the surface of the sphere and by r_g its radius. ψ^α varies with the scale of r_g near S_g for sufficiently large r_g , and the behavior at a far distance is well described by the Grad-Hilbert expansion [10] for small Kn/r_g ($\ll 1$). This implies that the flow velocity and the pressure due to ψ^α are described by the Stokes equation for the incompressible fluid, while the temperature due to ψ^α is described by the Laplace equation [10]. Thanks to the general solution of the Stokes equation [23–25] and the property of the harmonic functions (e.g., [26]), the pressure due to ψ^α is seen to be of $O(|\mathbf{x}|^{-2})$, while the flow velocity and temperature due to ψ^α are seen to be of $O(|\mathbf{x}|^{-1})$. Thus ψ^α for large $|\mathbf{x}|$ is estimated as

$$\psi^\alpha = 2\zeta_i c_i^\alpha + \left(|\boldsymbol{\zeta}|^2 - \frac{5}{2}\right)c^\alpha + O(|\mathbf{x}|^{-2}) \quad (\alpha = A, B),$$

where c_i^α and c^α are a quantity of $O(|\mathbf{x}|^{-1})$, independent of $\boldsymbol{\zeta}$, and we have

$$\int_{\partial D_g} \langle \zeta_n \psi^{A-} \psi^B \rangle dS = \lim_{r_g \rightarrow \infty} \int_{\partial S_g} \langle \zeta_n \psi^{A-} \psi^B \rangle dS = \lim_{r_g \rightarrow \infty} \int_{\partial S_g} O(|\mathbf{x}|^{-3}) dS = 0,$$

which is the desired equality. \square

3 Application to Half-Space Problems

We show some applications of the symmetric relation (14) to half-space problems. Throughout this section, the real boundary is assumed to be locally isotropic [10, 12]. Sections 3.1 and 3.2 present the recovery of known relations, while Sect. 3.3 provides unknown relations.

3.1 Shear and Thermal Creep Flows over a Plane Wall

When a rarefied gas is bounded by a wall with a gradient of temperature along its surface, a flow is induced along the wall in the direction of the gradient (the *thermal creep flow*; see, e.g., [1, 27–29]). We discuss a cross relation of this flow to the shear flow over the wall.

Let us denote by $x_1 L$ the space coordinate with $L = (\sqrt{\pi}/2)\ell_0$ where ℓ_0 is the mean free path of a molecule at the equilibrium state at rest with pressure p_0 and temperature T_0 . Consider a gas occupying a half space ($x_1 > 0$) over a resting plane wall in the following two situations depicted in Fig. 1:

1. *Thermal creep* (Problem T, for short). The temperature of the wall is given by $T_0(1 + C_T x_2)$ with C_T being a positive constant. At a far distance, the state of the gas is independent of x_1 , the pressure is p_0 , and the temperature is the same as that of the wall.
2. *Shear flow* (Problem S, for short). The temperature of the wall is T_0 . At a far distance, the gas pressure is p_0 , the temperature is T_0 , and the x_2 -component of the flow changes linearly in x_1 with a constant gradient $[(2kT_0/m)^{1/2}/L]C_S$, where k and m is the Boltzmann constant and the mass of a molecule, respectively.

⁴In general, h^α may have singularities in the domain under consideration, so does ψ^α . However, it does not cause any trouble, because only the outer region, where h^α and ψ^α are regular, will be considered.

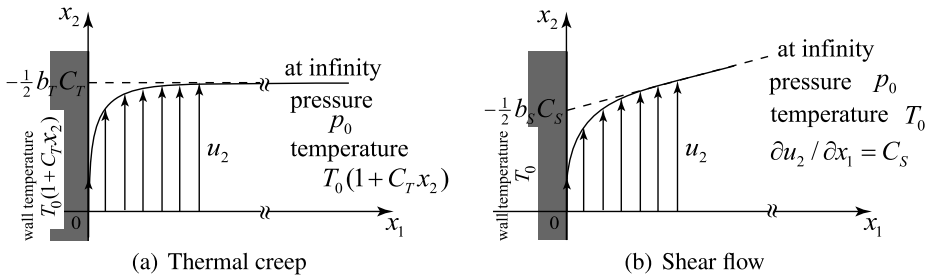


Fig. 1 Thermal creep and shear flow problems

The condition $C_T, |C_S| \ll 1$ is assumed for linearization. We denote by $2p_0(2kT_0/m)^{-5/2} \times (1 + \phi_J)E$ the velocity distribution function for problem J ($J = T, S$). ϕ_J is known to be sought in the form of $\phi_T = C_T[(|\xi|^2 - \frac{5}{2})x_2 - \Phi_T(x_1, \xi)]$ and $\phi_S = C_S[2\xi_2x_1 - \Phi_S(x_1, \xi)]$, where Φ_J is a solution of the following boundary-value problem:

$$\zeta_1 \frac{\partial \Phi_J}{\partial x_1} = \mathcal{L}(\Phi_J) + I_J \quad (J = T, S), \tag{15a}$$

$$\Phi_J = \int_{\zeta_1^* < 0} \frac{|\zeta_1^*|E(\zeta_1^*)}{\zeta_1 E(\zeta)} R_{CR}(\zeta^*, \xi) \phi_J^* d\zeta^*, \quad \zeta_1 > 0, \quad x_1 = 0, \tag{15b}$$

$$\Phi_J \rightarrow h_J \quad \text{as } x_1 \rightarrow \infty, \tag{15c}$$

where

$$I_T = \zeta_2 \left(|\xi|^2 - \frac{5}{2} \right), \quad h_T = b_T \zeta_2 + \zeta_2 A(|\xi|), \quad I_S = 2\zeta_1 \zeta_2, \quad h_S = b_S \zeta_2 + \zeta_1 \zeta_2 B(|\xi|),$$

R_{CR} is independent of \mathbf{x} , $A(|\xi|)$ is the solution of $\mathcal{L}(\zeta_i A) = -\zeta_i (|\xi|^2 - \frac{5}{2})$ such that $(|\xi|^2 A) = 0$, and $B(|\xi|)$ is the solution of $\mathcal{L}(\zeta_{ij} B) = -2\zeta_{ij}$ with $\zeta_{ij} = \zeta_i \zeta_j - \frac{1}{3} |\xi|^2 \delta_{ij}$.

The reduced problem is known to have a solution if and only if b_J takes a special value, and the solution is unique and approaches h_J exponentially fast as $x_1 \rightarrow \infty$ [22, 30–33]. With these properties in mind, we use Proposition 2 by putting $\phi^A = \phi_T$ and $\phi^B = \phi_S$ to have

$$-\frac{1}{2} \langle \zeta_1 h_S^- h_T \rangle - \int_0^\infty \langle I_S^- \Phi_T \rangle dx_1 = -\frac{1}{2} \langle \zeta_1 h_T^- h_S \rangle - \int_0^\infty \langle I_T^- \Phi_S \rangle dx_1.$$

In the meantime, the integration of (15a) for $J = T$ multiplied by $\zeta_2 E$ shows the second term on the left-hand side to vanish, and we eventually obtain

$$\langle \zeta_1^2 \zeta_2^2 B(|\xi|) \rangle b_T = - \int_0^\infty \left\langle \zeta_2 \left(|\xi|^2 - \frac{5}{2} \right) \Phi_S \right\rangle dx_1 - \langle \zeta_1^2 \zeta_2^2 A(|\xi|) B(|\xi|) \rangle. \tag{16}$$

As depicted in Fig. 1(a), $-\frac{1}{2} b_T$ is the dimensionless flow velocity at a far distance in the x_2 -direction normalized by C_T in problem T, while $\langle \zeta_2 (|\xi|^2 - \frac{5}{2}) \Phi_S \rangle$ is the dimensionless heat flow in the same direction normalized by C_S in problem S. Equation (16) represents the cross relation between the two problems. Incidentally, $2\langle \zeta_1^2 \zeta_2^2 B(|\xi|) \rangle$ is the dimensionless viscosity and $2\langle \zeta_1^2 \zeta_2^2 A(|\xi|) B(|\xi|) \rangle$ the dimensionless coefficient of the thermal stress.

The relation (16) can be checked by the numerical data in the literature (see, e.g., [20]). By using the notation in [10], (16) is rewritten as

$$\gamma_1 K_1 = -\frac{1}{2}\gamma_3 + \int_0^\infty H_A(\eta)d\eta, \tag{17}$$

where $\gamma_1 = 2\langle \zeta_1^2 \zeta_2^2 B \rangle$, $K_1 = b_T/2$, $\gamma_3 = 2\langle \zeta_1^2 \zeta_2^2 A B \rangle$, $H_A = -\langle \zeta_2(|\zeta|^2 - \frac{5}{2})\Phi_S \rangle$, and $\eta = x_1$. For the diffuse reflection boundary condition, K_1 and H_A have been computed accurately as

$$K_1 = -0.38316, \quad \int_0^\infty H_A(\eta)d\eta = \frac{1}{2} \cdot 0.23368 \quad (\text{BGK model [10, 27, 34]}),$$

$$K_1 = -0.6465, \quad \int_0^\infty H_A(\eta)d\eta = 0.1530 \quad (\text{Boltzmann equation; hard sphere [10, 28]}).$$

For the latter, we show the values recomputed from the data obtained in [35] ($K_1 = -0.6463$ and $\int_0^\infty H_A d\eta$ is not given in [10, 28]). According to [10], both γ_1 and γ_3 are unity for the BGK⁵ model, while $\gamma_1 = 1.270042427$ and $\gamma_3 = 1.947906335$ for the hard-sphere Boltzmann equation. Substitution of these values shows the relation (16) or (17) to hold.

3.2 Evaporation and Condensation on a Plane Condensed Phase

Consider a gas occupying a half space bounded by its plane condensed phase. The condensed phase is at rest and its temperature is uniform and constant. At a far distance, the gas is in the state of uniform pressure with uniform flow and temperature gradient normal to the interface. We discuss the steady behavior of the gas in the situation (see, e.g., [36–38]).

We use the same notation as Sect. 3.1 and take the temperature of the condensed phase and the corresponding saturation gas pressure as the reference temperature T_0 and pressure p_0 . If we denote by $2p_0(2kT_0/m)^{-5/2}(1 + \phi)E$ the velocity distribution function and by $T_0(1 + \Delta\tau + C_T x_1)$, $p_0(1 + \Delta P)$, and $(2kT_0/m)^{1/2}(u_\infty, 0, 0)$ the temperature, pressure, and flow velocity at a far distance, the problem is formulated as follows:

$$\zeta_1 \frac{\partial \phi}{\partial x_1} = \mathcal{L}(\phi),$$

$$\phi = \int_{\zeta_1^* < 0} \frac{|\zeta_1^*| E(\zeta^*)}{\zeta_1 E(\zeta)} R_{PR}(\zeta^*, \zeta) \phi^* d\zeta^*, \quad \zeta_1 > 0, \quad x_1 = 0,$$

$$\phi \rightarrow h \equiv \Delta P + \left(|\zeta|^2 - \frac{5}{2}\right)\Delta\tau + \left[\left(|\zeta|^2 - \frac{5}{2}\right)x_1 - \zeta_1 A(|\zeta|)\right]C_T + 2\zeta_1 u_\infty, \quad \text{as } x_1 \rightarrow \infty,$$

where R_{PR} is independent of \mathbf{x} and ΔP , $\Delta\tau$, C_T , and u_∞ are constants. Note that the heat flow $p_0(2kT_0/m)^{1/2}(Q_\infty, 0, 0)$ at a far distance is given by

$$Q_\infty = \left\langle \zeta_1 \left(|\zeta|^2 - \frac{5}{2}\right) \phi \right\rangle_{x_1 \rightarrow \infty} = \left\langle \zeta_1 \left(|\zeta|^2 - \frac{5}{2}\right) h \right\rangle = -\frac{1}{3} \langle |\zeta|^4 A(|\zeta|) \rangle C_T.$$

⁵The BGK (Bhatnagar–Gross–Krook) model is termed “BKW” (Boltzmann–Krook–Welander) equation in the cited references because of the independent contribution by Welander.

Denoting by subscript $(1, 0)$ the quantities $\phi, h, \Delta P,$ and $\Delta \tau$ when $(Q_\infty, u_\infty) = (1, 0)$ and by subscript $(0, 1)$ the counterparts when $(Q_\infty, u_\infty) = (0, 1)$, we can split the problem as

$$\begin{aligned} \phi &= \phi_{(1,0)} Q_\infty + \phi_{(0,1)} u_\infty, & h &= h_{(1,0)} Q_\infty + h_{(0,1)} u_\infty, \\ \Delta P &= \Delta P_{(1,0)} Q_\infty + \Delta P_{(0,1)} u_\infty, & \Delta T &= \Delta T_{(1,0)} Q_\infty + \Delta T_{(0,1)} u_\infty. \end{aligned}$$

As in Sect. 3.1, each of the reduced problems, the case of $(Q_\infty, u_\infty) = (1, 0)$ or $(0, 1)$, is known to have a solution if and only if $\Delta P_{(1,0)}$ and $\Delta \tau_{(1,0)}$ (or $\Delta P_{(0,1)}$ and $\Delta \tau_{(0,1)}$) take a special set of values, and the solution $\phi_{(0,1)}$ (or $\phi_{(1,0)}$) is unique and approaches $h_{(0,1)}$ (or $h_{(1,0)}$) exponentially fast as $x_1 \rightarrow \infty$.⁶ With these properties in mind, we make use of Proposition 2 with $\phi^A = \phi_{(1,0)}$ and $\phi^B = \phi_{(0,1)}$ to have the relation⁷

$$-\frac{1}{2} \langle \zeta_1 h_{(0,1)}^- h_{(1,0)} \rangle_{x_1 \rightarrow \infty} = -\frac{1}{2} \langle \zeta_1 h_{(1,0)}^- h_{(0,1)} \rangle_{x_1 \rightarrow \infty},$$

which is reduced to

$$\Delta P_{(1,0)} = \Delta \tau_{(0,1)}. \tag{18}$$

The relation (18) can be checked by numerical data in the literature. In fact, $\Delta P_{(1,0)}$ and $\Delta \tau_{(0,1)}$ are related to C_1 and d_4^* in [10] as $C_1 = -\frac{1}{3} \langle |\zeta|^4 A(|\zeta|) \rangle \Delta P_{(1,0)}$ and $d_4^* = \Delta \tau_{(0,1)}$, where C_1 and d_4^* for the complete condensation condition are given as

$$\begin{aligned} C_1 &= 0.55844, & d_4^* &= -0.44675 \quad (\text{BGK model [10, 34]}), \\ C_1 &= 1.0947, & d_4^* &= -0.4557 \quad (\text{Boltzmann equation; hard sphere [10, 37, 39]}). \end{aligned}$$

Note that $\frac{1}{3} \langle |\zeta|^4 A \rangle$ is $\frac{5}{2} \gamma_2$ in [10] ($\gamma_2 = 1$ for the BGK model and $\gamma_2 = 1.922284066$ for the hard-sphere Boltzmann equation). Substituting these values shows the relation (18) to hold.

3.3 Jump Condition for the Stokes Set of Equations on the Condensed Phase

Consider the steady behavior of a slightly rarefied gas around its condensed phase with arbitrary (smooth) shape in the linear regime. According to the asymptotic theory [10–12, 34] for small Knudsen numbers, the overall behavior of the gas can be described fluid-dynamically by the Stokes set of equations with a proper set of slip condition for the flow velocity and jump condition for the pressure and temperature. However, a correction is required in a thin layer adjacent to the interface with the thickness of a few mean free paths. The thin layer

⁶As is seen from the form of the equation and the boundary condition, $\phi_{(0,1)}$ (or $\phi_{(1,0)}$) can be sought as a function of $x_1, \zeta_1,$ and $|\zeta|$. This property will be used later in Sect. 3.3.

⁷There is a simple analogue in the theory of radiative transfer [18], in which the following two problems in a half space of isotropic scattering field are considered (F. Golse, private communication):

$$\begin{aligned} \mu \partial_x f &= -f + \langle f \rangle, & f(x=0, \mu > 0) &= \varphi(\mu), & f &\rightarrow f_\infty = \langle f_\infty \rangle \quad \text{as } x \rightarrow \infty, \\ \mu \partial_x g &= -g + \langle g \rangle, & g(x=0, \mu > 0) &= 0, & \langle \mu g \rangle &= -1, \end{aligned}$$

where $f(x, \mu)$ and $g(x, \mu)$ ($x \geq 0, -1 \leq \mu \leq 1$), the analogue to $\phi_{(0,1)}$ and $\phi_{(1,0)}$, denote the intensity of radiation, $\langle f \rangle = \frac{1}{2} \int_{-1}^1 f(x, \mu') d\mu'$, and φ is a given function. Here, concerned is mainly the value of f_∞ in the first problem and $g(x=0, \mu < 0)$ in the second. We can show that $f_\infty = \frac{1}{2} \int_0^1 \varphi(\mu') g(0, -\mu') d\mu'$ in a way similar to the derivation of (10), where $g(x=0, \mu > 0) = 0$ is essential to remove the contribution of $f(x=0, \mu < 0)$ from the identity. In this simple analogue, elaborate considerations on the boundary like in Sect. 2 are not necessary.

is called the Knudsen layer, and the correction is correspondingly called the Knudsen-layer correction; the solution of the Stokes set is called the fluid-dynamic part (of the solution of the Boltzmann equation). The study of the Knudsen layer is reduced to several half space problems of the homogeneous or inhomogeneous Boltzmann equation. The problems treated in Sects. 3.1 and 3.2 are typical examples of those reduced problems. By the analyses of the reduced problems, we can obtain the concrete value of the coefficients, the so-called slip and jump coefficients, occurring in the slip and jump conditions. For the details of the asymptotic theory, the reader is referred to [10, 12]. Here we merely present the general expressions for the pressure and jump conditions derived by the theory. We are going to show some relations among the jump coefficients obtained by the use of the symmetric relation (14).

Let us denote by τ and P the perturbed temperature and pressure of the fluid-dynamic part and by τ_w and P_w the perturbed temperature of the condensed phase and the corresponding perturbed saturation pressure of the gas. According to [10], if neglecting the second and higher order effects of the Knudsen number, the difference of τ and P from τ_w and P_w at the interface, which is called the temperature jump and pressure jump respectively, is given by

$$\begin{bmatrix} P - P_w \\ \tau - \tau_w \end{bmatrix} = u_n \begin{bmatrix} C_4^* \\ d_4^* \end{bmatrix} + \frac{\sqrt{\pi}}{2} \text{Kn} \left\{ \frac{\partial \tau}{\partial x_n} \begin{bmatrix} C_1 \\ d_1 \end{bmatrix} + 2 \frac{\partial u_n}{\partial x_n} \begin{bmatrix} C_6 \\ d_6 \end{bmatrix} - 2 \bar{\kappa} u_n \begin{bmatrix} C_7 \\ d_7 \end{bmatrix} \right\}.$$

Here u_n is the component inward normal to the interface of the fluid-dynamic part of the dimensionless flow velocity, x_n the same component of the dimensionless position vector, and $\bar{\kappa}/L$ is the mean curvature of the interface.⁸ In the above expression, u_n , $\partial \tau / \partial x_n$, and $\partial u_n / \partial x_n$ on the right-hand side represent their value at the interface.

As partially mentioned in Sect. 3.2, the coefficients (C_1, d_1) and (C_4^*, d_4^*) are related to ΔP 's and $\Delta \tau$'s there as $(C_1, d_1) = -\frac{5}{2} \gamma_2 (\Delta P_{(1,0)}, \Delta \tau_{(1,0)})$ and $(C_4^*, d_4^*) = (\Delta P_{(0,1)}, \Delta \tau_{(0,1)})$. On the other hand, (C_6, d_6) and (C_7, d_7) are respectively determined with the solution ϕ_6 and ϕ_7 of the following half space problems:

$$\zeta_1 \frac{\partial \phi_J}{\partial x_1} = \mathcal{L}(\phi_J) + I_J \quad (J = 6, 7), \tag{19a}$$

$$\phi_J = \int_{\zeta_1^* < 0} \frac{|\zeta_1^*| E(\zeta^*)}{\zeta_1 E(\zeta)} R_{\text{PR}}(\zeta^*, \zeta) \phi_J^* d\zeta^*, \quad \zeta_1 > 0, \quad x_1 = 0, \tag{19b}$$

$$\phi_J \rightarrow h_J, \quad \text{as } x_1 \rightarrow \infty, \tag{19c}$$

where

$$I_6 = \frac{1}{2} (3\zeta_1^2 - |\zeta|^2), \quad h_6 = -C_6 - \left(|\zeta|^2 - \frac{5}{2} \right) d_6 + \frac{1}{4} (3\zeta_1^2 - |\zeta|^2) B(|\zeta|),$$

$$I_7 = \frac{1}{2} (\zeta_1^2 - |\zeta|^2) \frac{\partial \psi_{(0,1)}}{\partial \zeta_1}, \quad h_7 = -C_7 - \left(|\zeta|^2 - \frac{5}{2} \right) d_7,$$

and R_{PR} is independent of \mathbf{x} . The $\psi_{(0,1)}$ is a function of x_1, ζ_1 , and $|\zeta|$ defined by $\psi_{(0,1)} = \phi_{(0,1)} - h_{(0,1)}$ (see the footnote 6) and vanishes exponentially as $x_1 \rightarrow \infty$.

⁸Here and in [10], the curvature is negative when the corresponding center of curvature lies on the gas side.

We first apply Proposition 2 to the pairs of ϕ_6 and $\phi_{(1,0)}$ by putting $\phi^A = \phi_6$ and $\phi^B = \phi_{(1,0)}$ and of ϕ_6 and $\phi_{(0,1)}$ by putting $\phi^A = \phi_6$ and $\phi^B = \phi_{(0,1)}$. Then we eventually obtain

$$C_6 = \gamma_1 - \frac{3}{4} \int_0^\infty (\Omega_4^* + \Theta_4^*) dx_1, \quad d_6 = \frac{1}{5\gamma_2} \left(2\gamma_3 + 3 \int_0^\infty (\Omega_1 + \Theta_1) dx_1 \right). \tag{20}$$

Here we followed the notation in [10], i.e., $\Omega_4^* + \Theta_4^* = \frac{2}{3} \langle |\xi|^2 \psi_{(0,1)} \rangle$, $\Omega_1 + \Theta_1 = -\frac{5}{6} \gamma_2 \langle |\xi|^2 \psi_{(1,0)} \rangle$, where $\gamma_1 \sim \gamma_3$ are those in Sects. 3.1 and 3.2 and $\psi_{(1,0)} = \phi_{(1,0)} - h_{(1,0)}$. The relation (20) can be checked by the numerical data in the literature. In the case of the complete condensation condition, they are given for the BGK model as follows [10, 34]:

$$\begin{aligned} C_6 &= 0.82085, & \int_0^\infty (\Omega_4^* + \Theta_4^*) dx_1 &= 0.23886, \\ d_6 &= 0.33034, & \int_0^\infty (\Omega_1 + \Theta_1) dx_1 &= -0.11609. \end{aligned}$$

Since $\gamma_1 = \gamma_2 = \gamma_3 = 1$ for this model, the relation (20) is seen to hold. On the other hand, the values of C_6 and d_6 have been unknown for the other cases, and here we report their values for the hard-sphere Boltzmann equation by the use of (20) with the aid of the data of $\psi_{(1,0)}$ and $\psi_{(0,1)}$ available in the literature (e.g., [37, 39–41]). By the use of the numerical data in [40, 41], the integrals of $\Omega_4^* + \Theta_4^*$ and $\Omega_1 + \Theta_1$ are computed as

$$\int_0^\infty (\Omega_4^* + \Theta_4^*) dx_1 = 0.2834, \quad \int_0^\infty (\Omega_1 + \Theta_1) dx_1 = -0.1943.$$

Since $\gamma_1 = 1.270042427$, $\gamma_2 = 1.922284066$, and $\gamma_3 = 1.947906335$, we have

$$C_6 = 1.0575, \quad d_6 = 0.3447 \quad (\text{Boltzmann equation; hard-sphere})$$

in the case of the complete condensation condition. To our best knowledge, this is the first report on the values of these coefficients for the hard-sphere Boltzmann equation.

Next we apply Proposition 2 to the pairs of ϕ_7 and $\phi_{(1,0)}$ by putting $\phi^A = \phi_7$ and $\phi^B = \phi_{(1,0)}$ and of ϕ_7 and $\phi_{(0,1)}$ by putting $\phi^A = \phi_7$ and $\phi^B = \phi_{(0,1)}$. Then, we eventually obtain

$$C_7 = \frac{1}{2} \int_0^\infty \left\langle (\xi_1^2 - |\xi|^2) \psi_{(0,1)}^- \frac{\partial \psi_{(0,1)}}{\partial \xi_1} \right\rangle dx_1 - \frac{3}{2} \int_0^\infty (\Omega_4^* + \Theta_4^*) dx_1, \tag{21a}$$

$$d_7 = \frac{1}{2} \int_0^\infty \left\langle (\xi_1^2 - |\xi|^2) \psi_{(1,0)}^- \frac{\partial \psi_{(0,1)}}{\partial \xi_1} \right\rangle dx_1 + \frac{1}{5\gamma_2} \int_0^\infty \langle (3\xi_1^2 - |\xi|^2) A(|\xi|) \psi_{(0,1)} \rangle dx_1. \tag{21b}$$

The present example demonstrates the advantage of a rather abstract formulation in Sect. 2. The problem for ϕ_7 contains the derivative of non Chapman–Enskog solution $\psi_{(0,1)}$ as the inhomogeneous term, which is typical in the higher-order analyses of the Knudsen-layer. The feature that I (and h) is not required to be of a specific form is advantageous in such analyses and will allow us to derive further identities for slip and jump coefficients.

4 Representation Theorem on Mass, Momentum, and Heat Fluxes

As noted just before Lemma 2 in Sect. 2.2.2, a remarkable difference of the bounded and three dimensional unbounded domains from the half space is the fact that one may separately

discuss the effect of the sources on the boundary. We shall present a fruitful consequence obtained from this property linked to the symmetric relation (14).

4.1 Problem and Formulation

Consider the steady behavior of a rarefied gas in a domain that is arbitrary except for the conditions to be described later. There is no external force. The state of the gas is so close to the reference equilibrium state at rest with density ρ_0 and temperature T_0 that the higher order terms of the deviation from the reference state may be neglected. We shall mainly discuss the general properties of the mass, momentum, and heat transferred to the boundary of the domain under consideration.

Let us denote by L the reference length of the system, by $L\mathbf{x}$ the position, by $(2kT_0/m)^{1/2}\boldsymbol{\zeta}$ the molecular velocity, and by $\rho_0(2kT_0/m)^{-3/2}(1 + \phi)E(\boldsymbol{\zeta})$ the velocity distribution function of gas molecules, where k is the Boltzmann constant and m is the mass of a gas molecule. Then, the behavior of the gas, the perturbed velocity distribution function ϕ , is described by the following linearized Boltzmann equation:

$$\zeta_i \frac{\partial \phi}{\partial x_i} = \frac{2}{\sqrt{\pi}} \frac{1}{\text{Kn}} \mathcal{L}(\phi), \tag{22}$$

where Kn is the Knudsen number defined by $\text{Kn} = \ell_0/L$ with ℓ_0 being the mean free path of a molecule in the gas at the reference equilibrium state.

Let us denote by D the domain of \mathbf{x} representing the gas domain under consideration. The boundary ∂D of the domain is split into two parts: the part ∂D_w representing the *real boundary* and the part ∂D_g representing the remaining *imaginary boundary*. As the domain D , we consider the following two possibilities:

1. D is a bounded domain.
2. D is an unbounded domain, and the real boundary ∂D_w is confined in a finite region, i.e., there exists a sphere with a finite radius that contains ∂D_w in its interior.

It is important to note that the second assumption excludes unbounded domains in one and two-dimensional problems from the subsequent discussions.

As to the real boundary part, we denote by $T_0(1 + \tau_w)$ the temperature of the real boundary ∂D_w , by $p_0(1 + P_w)$ the corresponding saturation pressure of the gas, and by $(2kT_0/m)^{1/2}\mathbf{u}_w$ the velocity of the boundary, where $p_0 = (\rho_0/m)kT_0$. Then, ϕ obeys the following condition on ∂D_w :

$$\phi = g_w + \int_{\zeta_n^* < 0} \frac{|\zeta_n^*| E(\boldsymbol{\zeta}^*)}{|\zeta_n| E(\boldsymbol{\zeta})} R(\boldsymbol{\zeta}^*, \boldsymbol{\zeta}; \mathbf{x})(\phi^* - g_w^*) d\boldsymbol{\zeta}^* \quad \text{for } \zeta_n > 0. \tag{23}$$

Here $R = R_{\text{CR}}$ on the simple boundary and $R = R_{\text{PR}}$ on the interface, where R_{CR} and R_{PR} are those in Appendix A that satisfy the condition of detailed balance (31). The function g_w is given by (5). The kernel R multiplied by $(2kT_0/m)^{3/2}$ is the reflection kernel of the real boundary which is at rest with the reference temperature T_0 . In what follows, irrespective of the type of the real boundary (either $R = R_{\text{CR}}$ or R_{PR}), we simply write g_w as

$$g_w = P_w + 2\zeta_i u_{wi} + \left(|\boldsymbol{\zeta}|^2 - \frac{5}{2} \right) \tau_w \quad \text{on } \partial D_w, \tag{24}$$

because the addition of any $\boldsymbol{\zeta}$ -independent term to g_w does not influence the condition (23) because of the third property of R_{CR} in Appendix A. Note that $u_{wi} n_i = 0$ because we are concerned with the steady problem.

On the imaginary boundary ∂D_g , ϕ obeys the following condition:

1. When D is bounded, $\partial D_g = \partial D_g^{(i)} \cup \partial D_g^{(iii)}$ and ϕ obeys the condition (11a) or (11c):

$$\phi(\mathbf{x}, \boldsymbol{\zeta}) = h_{in}(\mathbf{x}, \boldsymbol{\zeta}) \quad \text{for } \zeta_n > 0 \text{ on } \partial D_g^{(i)}, \tag{25a}$$

$$\phi(\mathbf{x}, \boldsymbol{\zeta}) = h(\mathbf{x}, \boldsymbol{\zeta}) + \int_{\partial D_g^{(iii)}} \int_{\zeta'_n < 0} P(\mathbf{x}', \boldsymbol{\zeta}', \mathbf{x}, \boldsymbol{\zeta})(\phi' - h') d\boldsymbol{\zeta}' dS' \quad \text{for } \zeta_n > 0 \text{ on } \partial D_g^{(iii)}. \tag{25b}$$

2. When D is unbounded, $\partial D_g = \partial D_g^{(ii)}$, and ϕ obeys the condition (11b):

$$\phi(\mathbf{x}, \boldsymbol{\zeta}) \rightarrow h(\mathbf{x}, \boldsymbol{\zeta}) \quad \text{as } |\mathbf{x}| \rightarrow \infty, \tag{25c}$$

where h is a solution of (22).

Note that h is defined for the whole range of $\boldsymbol{\zeta}$ in each case. As in Sect. 2.2.1, we extend the range of the function h_{in} from $\zeta_n > 0$ to the whole range of $\boldsymbol{\zeta}$ and denote the extended function by h . The way of extension is arbitrary and does not affect the subsequent results.

Most of the cases, we are concerned with the flow velocity $(2kT_0/m)^{1/2}u_i$, heat-flow vector $\frac{1}{2}\rho_0(2kT_0/m)^{3/2}Q_i$, and stress tensor $p_0(\delta_{ij} + P_{ij})$ of the gas, where δ_{ij} is the Kronecker delta. In particular, their component normal to the boundary is of special interest, which will be denoted by u_n , Q_n , and P_{nj} :

$$u_n = u_i n_i = \langle \zeta_n \phi \rangle, \quad Q_n = Q_i n_i = \left\langle \zeta_n \left(|\boldsymbol{\zeta}|^2 - \frac{5}{2} \right) \phi \right\rangle, \quad P_{nj} = P_{ij} n_i = \langle 2\zeta_n \zeta_j \phi \rangle.$$

Note that they represent the dimensionless inward fluxes of mass, heat, and momentum through the boundary at the position, respectively.

As is noted in Sect. 2.2.2, all the conditions in Proposition 2 are fulfilled by any two solutions of (22), (23), and (25) with g_w given by (24). The first condition is obviously fulfilled by the definition of R , while the second is due to Lemma 2.⁹ Thus, the symmetric relation (14) holds between any two solutions discussed in Sect. 4.

4.2 Preliminary Argument—Motivation and Basic Results

If $h = 0$ on ∂D_g , the perturbation ϕ from the reference equilibrium state is induced only by the source g_w on ∂D_w . Since g_w is a linear combination of 1, $\boldsymbol{\zeta}$ with $\zeta_n = 0$, and $|\boldsymbol{\zeta}|^2 - \frac{5}{2}$, we are motivated to consider the response of the system to these elemental sources on ∂D_w and to represent ϕ as their superposition.

Let \mathbf{x}_0 be a point on the real boundary ∂D_w . We introduce three elemental solutions of the boundary-value problem (22), (23), and (25) listed in the first three lines in Table 1, which we denote by $G^{(P;\mathbf{x}_0)}(\mathbf{x}, \boldsymbol{\zeta})$, $G^{(T;\mathbf{x}_0)}(\mathbf{x}, \boldsymbol{\zeta})$, and $G^{(f;\mathbf{x}_0)}(\mathbf{x}, \boldsymbol{\zeta})$. Each of them represents the response of the system to the corresponding elemental source put on ∂D_w , and thus we call

⁹Note that the addition of P_w to g_w made in (24) on a simple boundary does not change the value of the first term of each side of (14) because $\langle \zeta_n \phi \rangle = 0$ there.

Table 1 Green functions for elemental sources on ∂D_w . The domain may be bounded or unbounded

Green function ^a	Corresponding elemental source ^b		Note
$G^{(P;x_0)}(\mathbf{x}, \boldsymbol{\zeta})$	$g_w = \delta(\mathbf{x} - \mathbf{x}_0)$	$h = 0$	—
$G^{(T;x_0)}(\mathbf{x}, \boldsymbol{\zeta})$	$g_w = (\boldsymbol{\zeta} ^2 - \frac{5}{2})\delta(\mathbf{x} - \mathbf{x}_0)$	$h = 0$	—
$G^{(t;x_0)}(\mathbf{x}, \boldsymbol{\zeta})$	$g_w = 2\zeta_i t_i \delta(\mathbf{x} - \mathbf{x}_0)$	$h = 0$	$G^{(t;x_0)}(\mathbf{x}, \boldsymbol{\zeta}) = -G^{(-t;x_0)}(\mathbf{x}, \boldsymbol{\zeta})$
$G^{(P;S)}(\mathbf{x}, \boldsymbol{\zeta})$	$g_w = \chi_S \delta(\mathbf{x})$	$h = 0$	$G^{(P;S)}(\mathbf{x}, \boldsymbol{\zeta}) = \int_S G^{(P;x_0)}(\mathbf{x}, \boldsymbol{\zeta}) dS_0$
$G^{(T;S)}(\mathbf{x}, \boldsymbol{\zeta})$	$g_w = (\boldsymbol{\zeta} ^2 - \frac{5}{2})\chi_S(\mathbf{x})$	$h = 0$	$G^{(T;S)}(\mathbf{x}, \boldsymbol{\zeta}) = \int_S G^{(T;x_0)}(\mathbf{x}, \boldsymbol{\zeta}) dS_0$

^a \mathbf{t} is a unit vector tangential to ∂D_w at \mathbf{x}_0 . $S \subseteq \partial D_w$

^b δ is the δ -function. $\chi_S(\mathbf{x}) = 1$ for $\mathbf{x} \in S$ and $\chi_S(\mathbf{x}) = 0$ otherwise

them the *Green functions*¹⁰ with respect to \mathbf{x}_0 on ∂D_w . If necessary, we call $G^{(P;x_0)}$, $G^{(T;x_0)}$, and $G^{(t;x_0)}$ the Green function for the pressure source, temperature source, and velocity source in the \mathbf{t} -direction, respectively. Any solution ϕ of the boundary-value problem (22), (23), and (25) with $h = 0$ can be expressed as a superposition of the Green functions:

$$\phi = \int_{\partial D_w} (P_w(\mathbf{x}_0)G^{(P;x_0)} + \tau_w(\mathbf{x}_0)G^{(T;x_0)} - |\mathbf{u}_w(\mathbf{x}_0)|G^{(-\hat{\mathbf{u}}_w;x_0)})dS_0, \tag{26}$$

where $\hat{\mathbf{u}}_w = \mathbf{u}_w/|\mathbf{u}_w|$ and dS_0 is the surface element at position \mathbf{x}_0 . Here and in what follows, we denote the moments of the Green function by putting the corresponding superscript. For instance, $u_i^{(P;x_0)}(\mathbf{x}) = \langle \zeta_i G^{(P;x_0)}(\mathbf{x}, \boldsymbol{\zeta}) \rangle$. The first important observation is a reciprocity of the Green functions in the following sense:

Lemma 3 For any points \mathbf{x}_0 and \mathbf{x}_1 on ∂D_w , the following relation holds:

$$\begin{aligned} & \begin{bmatrix} u_n^{(P;x_1)}(\mathbf{x}_0), & u_n^{(T;x_1)}(\mathbf{x}_0), & u_n^{(-s;x_1)}(\mathbf{x}_0) \\ Q_n^{(P;x_1)}(\mathbf{x}_0), & Q_n^{(T;x_1)}(\mathbf{x}_0), & Q_n^{(-s;x_1)}(\mathbf{x}_0) \\ P_{nt}^{(P;x_1)}(\mathbf{x}_0), & P_{nt}^{(T;x_1)}(\mathbf{x}_0), & P_{nt}^{(-s;x_1)}(\mathbf{x}_0) \end{bmatrix} \\ &= \begin{bmatrix} u_n^{(P;x_0)}(\mathbf{x}_1), & Q_n^{(P;x_0)}(\mathbf{x}_1), & P_{ns}^{(P;x_0)}(\mathbf{x}_1) \\ u_n^{(T;x_0)}(\mathbf{x}_1), & Q_n^{(T;x_0)}(\mathbf{x}_1), & P_{ns}^{(T;x_0)}(\mathbf{x}_1) \\ u_n^{(-t;x_0)}(\mathbf{x}_1), & Q_n^{(-t;x_0)}(\mathbf{x}_1), & P_{ns}^{(-t;x_0)}(\mathbf{x}_1) \end{bmatrix}, \end{aligned} \tag{27}$$

where $P_{nt} = P_{nj}t_j$, $P_{ns} = P_{nj}s_j$, and \mathbf{t} and \mathbf{s} are a unit vector tangential to the boundary at point \mathbf{x}_0 and \mathbf{x}_1 respectively.

Proof Proposition 2 is applicable to any two Green functions (see the last paragraph of Sect. 4.1). With $\phi^A = G^{(\alpha;x_0)}$ and $\phi^B = G^{(\beta;x_1)}$ ($\alpha = P, T, -\mathbf{t}$; $\beta = P, T, -\mathbf{s}$), we apply the symmetric relation (14). Since $I^A = 0$, $I^B = 0$, $h^A = 0$, and $h^B = 0$, (14) is reduced to

$$\int_{\partial D_w} \langle \zeta_n g_w^{A-} G^{(\beta;x_1)}(\mathbf{x}, \boldsymbol{\zeta}) \rangle dS = \int_{\partial D_w} \langle \zeta_n g_w^{B-} G^{(\alpha;x_0)}(\mathbf{x}, \boldsymbol{\zeta}) \rangle dS.$$

¹⁰In the present paper, we shall use the term ‘‘Green functions’’ for the solution even when the corresponding elemental source is not a point source. See, for instance, the last two lines in Table 1.

Let us consider the case $\alpha = P$ and $\beta = T$. Since $g_w^A = \delta(\mathbf{x} - \mathbf{x}_0)$ and $g_w^B = (|\zeta|^2 - \frac{5}{2})\delta(\mathbf{x} - \mathbf{x}_1)$, performing the surface integration on ∂D_w yields

$$\langle \zeta_n G^{(T;\mathbf{x}_1)}(\mathbf{x}_0, \zeta) \rangle = \left\langle \zeta_n \left(|\zeta|^2 - \frac{5}{2} \right) G^{(P;\mathbf{x}_0)}(\mathbf{x}_1, \zeta) \right\rangle,$$

which is no other than the equality $u_n^{(T;\mathbf{x}_1)}(\mathbf{x}_0) = Q_n^{(P;\mathbf{x}_0)}(\mathbf{x}_1)$. Other equalities can be proved in the same way by changing the pair of α and β . □

Lemma 3 leads to the following important consequence, if it is linked to (26):

Corollary 1 Consider the boundary-value problem (22), (23), and (25) with $h = 0$. For any point \mathbf{x}_0 on ∂D_w , the fluxes normal to the boundary can be expressed by

$$\begin{bmatrix} u_n(\mathbf{x}_0) \\ Q_n(\mathbf{x}_0) \\ P_{ni}(\mathbf{x}_0) \end{bmatrix} = \int_{\partial D_w} \begin{bmatrix} P_w(\mathbf{x})u_n^{(P;\mathbf{x}_0)}(\mathbf{x}) + \tau_w(\mathbf{x})Q_n^{(P;\mathbf{x}_0)}(\mathbf{x}) - u_{wj}(\mathbf{x})P_{nj}^{(P;\mathbf{x}_0)}(\mathbf{x}) \\ P_w(\mathbf{x})u_n^{(T;\mathbf{x}_0)}(\mathbf{x}) + \tau_w(\mathbf{x})Q_n^{(T;\mathbf{x}_0)}(\mathbf{x}) - u_{wj}(\mathbf{x})P_{nj}^{(T;\mathbf{x}_0)}(\mathbf{x}) \\ P_w(\mathbf{x})u_n^{(-t;\mathbf{x}_0)}(\mathbf{x}) + \tau_w(\mathbf{x})Q_n^{(-t;\mathbf{x}_0)}(\mathbf{x}) - u_{wj}(\mathbf{x})P_{nj}^{(-t;\mathbf{x}_0)}(\mathbf{x}) \end{bmatrix} dS, \tag{28}$$

where \mathbf{t} is a unit vector tangential to the boundary at \mathbf{x}_0 .

Proof The superposition (26) yields

$$\begin{bmatrix} u_n(\mathbf{x}_0) \\ Q_n(\mathbf{x}_0) \\ P_{ni}(\mathbf{x}_0) \end{bmatrix} = \int_{\partial D_w} \begin{bmatrix} P_w(\mathbf{x})u_n^{(P;\mathbf{x})}(\mathbf{x}_0) + \tau_w(\mathbf{x})u_n^{(T;\mathbf{x})}(\mathbf{x}_0) - |\mathbf{u}_w(\mathbf{x})| u_n^{(-\hat{u}_w;\mathbf{x})}(\mathbf{x}_0) \\ P_w(\mathbf{x})Q_n^{(P;\mathbf{x})}(\mathbf{x}_0) + \tau_w(\mathbf{x})Q_n^{(T;\mathbf{x})}(\mathbf{x}_0) - |\mathbf{u}_w(\mathbf{x})| Q_n^{(-\hat{u}_w;\mathbf{x})}(\mathbf{x}_0) \\ P_w(\mathbf{x})P_{ni}^{(P;\mathbf{x})}(\mathbf{x}_0) + \tau_w(\mathbf{x})P_{ni}^{(T;\mathbf{x})}(\mathbf{x}_0) - |\mathbf{u}_w(\mathbf{x})| P_{ni}^{(-\hat{u}_w;\mathbf{x})}(\mathbf{x}_0) \end{bmatrix} dS,$$

and the substitution of (27) leads to the desired expression. □

Corollary 1 shows that each of the mass, tangential momentum, and heat fluxes at a point \mathbf{x}_0 on the real boundary ∂D_w is expressed as a weighted sum of the fluxes over the boundary ∂D_w induced by the corresponding Green function with respect to that point. The correspondence is as follows: the mass flux \Leftrightarrow the pressure source, the heat flux \Leftrightarrow the temperature source, and the momentum flux \Leftrightarrow the velocity source.

4.3 Representation Theorem

Corollary 1 implies a reduction of the original problem to a single elemental problem of finding the Green function corresponding to the flux of interest. A natural question arises whether a similar representation could be obtained for general situations such that h is not necessarily zero. Fortunately, we can give an affirmative answer to this question, which we present here. The key to the generalization is the fact that one can recover Corollary 1 by directly applying Proposition 2 to the pair of ϕ in Sect. 4.2 and the Green function. To be specific, the representations of $u_n(\mathbf{x}_0)$, $Q_n(\mathbf{x}_0)$, $P_{ni}(\mathbf{x}_0)$ are respectively recovered by the application of Proposition 2 to the pairs of ϕ and $G^{(P;\mathbf{x}_0)}$, ϕ and $G^{(T;\mathbf{x}_0)}$, and ϕ and $G^{(-t;\mathbf{x}_0)}$. This alternative approach enables us to perform the generalization by the use of suitable elemental sources and the corresponding Green functions. We call the set of expressions thus obtained the *representation theorem on mass, momentum, and heat fluxes*.

Actually, there are five versions of the representation theorem depending on the situation. We present them in Sects. 4.3.1–4.3.3.

4.3.1 Representation of Fluxes Through ∂D_w

Proposition 3 (Representation Theorem 1) *Consider the boundary-value problem (22), (23), and (25). The fluxes of mass, heat, and tangential momentum through ∂D_w at \mathbf{x}_0 are represented in terms of the corresponding Green function, irrespective of whether D is bounded or unbounded:*

$$\begin{aligned} \begin{bmatrix} u_n(\mathbf{x}_0) \\ Q_n(\mathbf{x}_0) \\ P_{ni}(\mathbf{x}_0) \end{bmatrix} &= \int_{\partial D_w} \begin{bmatrix} P_w u_n^{(P;\mathbf{x}_0)} + \tau_w Q_n^{(P;\mathbf{x}_0)} - u_{wi} P_{ni}^{(P;\mathbf{x}_0)} \\ P_w u_n^{(T;\mathbf{x}_0)} + \tau_w Q_n^{(T;\mathbf{x}_0)} - u_{wi} P_{ni}^{(T;\mathbf{x}_0)} \\ P_w u_n^{(-t;\mathbf{x}_0)} + \tau_w Q_n^{(-t;\mathbf{x}_0)} - u_{wi} P_{ni}^{(-t;\mathbf{x}_0)} \end{bmatrix} dS \\ &+ \int_{\partial D_g} \begin{bmatrix} \langle \zeta_n h^- G^{(P;\mathbf{x}_0)} \rangle \\ \langle \zeta_n h^- G^{(T;\mathbf{x}_0)} \rangle \\ \langle \zeta_n h^- G^{(-t;\mathbf{x}_0)} \rangle \end{bmatrix} dS, \end{aligned}$$

where \mathbf{t} is an arbitrary unit vector tangential to the boundary at \mathbf{x}_0 .¹¹ The way of extension of h_{in} on $\partial D_g^{(i)}$ does not influence the relation.¹²

Proof We apply Proposition 2 to the pair of $\phi^B = G^{(\alpha;\mathbf{x}_0)}$ ($\alpha = P, T, -t$) and $\phi^A = \phi$, where ϕ is the solution of the boundary-value problem (22), (23), and (25). Since $I^A = 0$, $g_w^A = g_w$, $h^A = h$, $I^B = 0$, and $h^B = 0$, the symmetric relation (14) is reduced to

$$\int_{\partial D_w} \langle \zeta_n g_w^B \phi \rangle dS = \int_{\partial D_w} \langle \zeta_n g_w^- G^{(\alpha;\mathbf{x}_0)} \rangle dS + \int_{\partial D_g} \langle \zeta_n h^- G^{(\alpha;\mathbf{x}_0)} \rangle dS,$$

where $g_w^B = \delta(\mathbf{x} - \mathbf{x}_0)$, $(|\zeta|^2 - \frac{5}{2})\delta(\mathbf{x} - \mathbf{x}_0)$, and $-2\zeta_i t_i \delta(\mathbf{x} - \mathbf{x}_0)$ for $\alpha = P, T$, and t , respectively. Substitution of the specific form of g_w^B and g_w yields the desired representation. \square

Proposition 3 can be transformed into the statement on the fluxes through an arbitrary area on ∂D_w , which is useful in many applications. To derive it, we introduce the Green functions with respect to an area on ∂D_w defined in the last two lines in Table 1. If we denote by $\rho_0 L^2 (2kT_0/m)^{1/2} \mathcal{M}(A_w)$ and $\frac{1}{2} \rho_0 L^2 (2kT_0/m)^{3/2} \mathcal{Q}(A_w)$ the mass and heat transferred to an area A_w on ∂D_w per unit time, they are written in terms of u_n and Q_n as

$$\begin{bmatrix} \mathcal{M}(A_w) \\ \mathcal{Q}(A_w) \end{bmatrix} = - \int_{A_w} \begin{bmatrix} u_n(\mathbf{x}_0) \\ Q_n(\mathbf{x}_0) \end{bmatrix} dS_0.$$

Substituting the representation in Proposition 3 and performing the surface integration with respect to \mathbf{x}_0 leads to the representation in terms of the Green functions with respect to A_w .¹³

¹¹We may consider the Green function $G^{(\alpha;\mathbf{x}_0)}$ that is the solution of the problem (22), (23), and (25) with $h = 0$ and $g_w = 2\zeta_n \delta(\mathbf{x} - \mathbf{x}_0)$. Then, we can derive the representation for an arbitrary component of the force. In the present work, however, we restrict ourselves to derive a representation in terms of the Green functions satisfying the physical requirement $\mathbf{u}_w \cdot \mathbf{n} = 0$. The same is true for the general representations of the torque on the real boundary.

¹²On $\partial D_g^{(i)}$, h is an arbitrary extension of h_{in} from $\zeta_n > 0$ to the whole range of ζ , and hence h^- for $\zeta_n > 0$ is the extended part. The arbitrariness of this part is killed in the moment $\langle \zeta_n h^- G^{(\alpha;\mathbf{x}_0)} \rangle$ ($\alpha = P, T, -t$), because $G^{(\alpha;\mathbf{x}_0)} = 0$ for $\zeta_n > 0$ by definition.

¹³The same representation is obtained by applying Proposition 2 to the pair of $\phi^B = G^{(\alpha;A_w)}$ and $\phi^A = \phi$ directly, which would be a less demanding way in mathematical rigor.

Proposition 4 (Representation Theorem 2) *Consider the boundary-value problem (22), (23), and (25). Irrespective of whether D is bounded or unbounded, the outward fluxes of mass and heat through an area A_w on ∂D_w are represented in terms of the corresponding Green function with respect to this area:*

$$\begin{aligned} \begin{bmatrix} \mathcal{M}(A_w) \\ \mathcal{Q}(A_w) \end{bmatrix} &= - \int_{\partial D_w} \begin{bmatrix} P_w u_n^{(P;A_w)} + \tau_w Q_n^{(P;A_w)} - u_{wi} P_{ni}^{(P;A_w)} \\ P_w u_n^{(T;A_w)} + \tau_w Q_n^{(T;A_w)} - u_{wi} P_{ni}^{(T;A_w)} \end{bmatrix} dS \\ &\quad - \int_{\partial D_g} \begin{bmatrix} \langle \zeta_n h^- G^{(P;A_w)} \rangle \\ \langle \zeta_n h^- G^{(T;A_w)} \rangle \end{bmatrix} dS. \end{aligned}$$

The way of extension of h_{in} on $\partial D_g^{(i)}$ does not influence the relation (see footnote 12).¹⁴

4.3.2 Representation of Fluxes Through ∂D_g for a Bounded Domain

As to the fluxes through ∂D_g , we need to discuss the bounded and unbounded domains separately. Here, we focus on the former and introduce the Green functions listed in Table 2. The main difference from the previous case lies in the Green function for the velocity source, i.e., the direction of the velocity source ℓ is not necessarily tangential to the boundary but rather arbitrary. This feature allows us to have the representation of not only the mass and heat fluxes but also the momentum flux through an arbitrary area on ∂D_g . We first present the representation for fluxes through a point on ∂D_g . The proof is similar to that in Sect. 4.3.1 and is omitted for the sake of brevity.

Proposition 5 (Representation Theorem 3) *Consider the boundary-value problem (22), (23), and (25a) or (25b) for a bounded domain. The fluxes of mass, heat, and tangential momentum through ∂D_g at \mathbf{x}_g are represented in terms of the corresponding Green func-*

Table 2 Green functions for elemental sources on ∂D_g for a bounded domain

Green function ^a	Corresponding elemental source ^b	Note
$G^{(P;x_0)}(\mathbf{x}, \zeta)$	$g_w = 0 \quad h = \delta(\mathbf{x} - \mathbf{x}_0)$	—
$G^{(T;x_0)}(\mathbf{x}, \zeta)$	$g_w = 0 \quad h = (\zeta ^2 - \frac{5}{2})\delta(\mathbf{x} - \mathbf{x}_0)$	—
$G^{(\ell;x_0)}(\mathbf{x}, \zeta)$	$g_w = 0 \quad h = 2\zeta_i \ell_i \delta(\mathbf{x} - \mathbf{x}_0)$	$G^{(\ell;x_0)}(\mathbf{x}, \zeta) = -G^{(-\ell;x_0)}(\mathbf{x}, \zeta)$
$G^{(P;S)}(\mathbf{x}, \zeta)$	$g_w = 0 \quad h = \chi_S(\mathbf{x})$	$G^{(P;S)}(\mathbf{x}, \zeta) = \int_S G^{(P;x_0)}(\mathbf{x}, \zeta) dS_0$
$G^{(T;S)}(\mathbf{x}, \zeta)$	$g_w = 0 \quad h = (\zeta ^2 - \frac{5}{2})\chi_S(\mathbf{x})$	$G^{(T;S)}(\mathbf{x}, \zeta) = \int_S G^{(T;x_0)}(\mathbf{x}, \zeta) dS_0$
$G^{(\ell;S)}(\mathbf{x}, \zeta)$	$g_w = 0 \quad h = 2\zeta_i \ell_i \chi_S(\mathbf{x})$	$G^{(\ell;S)}(\mathbf{x}, \zeta) = \int_S G^{(\ell;x_0)}(\mathbf{x}, \zeta) dS_0$ $G^{(\ell;S)}(\mathbf{x}, \zeta) = -G^{(-\ell;S)}(\mathbf{x}, \zeta)$

^a ℓ is an arbitrary unit vector. $S \subseteq \partial D_g$

^bSee the footnote b in Table 1

¹⁴We exclude from the general discussion the possibility of the Green function for the velocity source, because, in general, the direction of the vector t tangential to the boundary depends on the position.

tion:

$$\begin{aligned} \begin{bmatrix} u_n(\mathbf{x}_g) \\ Q_n(\mathbf{x}_g) \\ P_{n\ell}(\mathbf{x}_g) \end{bmatrix} &= \int_{\partial D_w} \begin{bmatrix} P_w u_n^{(P;\mathbf{x}_g)} + \tau_w Q_n^{(P;\mathbf{x}_g)} - u_{wj} P_{nj}^{(P;\mathbf{x}_g)} \\ P_w u_n^{(T;\mathbf{x}_g)} + \tau_w Q_n^{(T;\mathbf{x}_g)} - u_{wj} P_{nj}^{(T;\mathbf{x}_g)} \\ P_w u_n^{(-\ell;\mathbf{x}_g)} + \tau_w Q_n^{(-\ell;\mathbf{x}_g)} - u_{wj} P_{nj}^{(-\ell;\mathbf{x}_g)} \end{bmatrix} dS \\ &+ \int_{\partial D_g} \begin{bmatrix} \langle \zeta_n h^- G^{(P;\mathbf{x}_g)} \rangle \\ \langle \zeta_n h^- G^{(T;\mathbf{x}_g)} \rangle \\ \langle \zeta_n h^- G^{(-\ell;\mathbf{x}_g)} \rangle \end{bmatrix} dS + \begin{bmatrix} \langle \zeta_n h(\mathbf{x}_g, \boldsymbol{\zeta}) \rangle \\ \langle \zeta_n (|\boldsymbol{\zeta}|^2 - \frac{5}{2}) h(\mathbf{x}_g, \boldsymbol{\zeta}) \rangle \\ \langle 2\zeta_n \zeta_j \ell_j h(\mathbf{x}_g, \boldsymbol{\zeta}) \rangle \end{bmatrix}, \end{aligned}$$

where $P_{n\ell} = P_{nj} \ell_j$ and ℓ is an arbitrary unit vector. The way of extension of h_{in} on $\partial D_g^{(i)}$ does not influence the relation.¹⁵

Next, we show the representation of the fluxes through an area on ∂D_g . Let us denote by $\rho_0 L^2 (2kT_0/m)^{1/2} \mathcal{M}(A_g)$, $\frac{1}{2} \rho_0 L^2 (2kT_0/m)^{3/2} \mathcal{Q}(A_g)$, and $p_0 L^2 \mathcal{F}_\ell(A_g)$ the mass, heat, and momentum in the ℓ -direction transferred to the area A_g on ∂D_g for a bounded domain per unit time. $p_0 L^2 \mathcal{F}_\ell(A_g)$ may be regarded as the force acting on the area A_g in the ℓ -direction. They are written in terms of u_n , Q_n and $P_{n\ell}$ as

$$\begin{bmatrix} \mathcal{M}(A_g) \\ \mathcal{Q}(A_g) \\ \mathcal{F}_\ell(A_g) \end{bmatrix} = - \int_{A_g} \begin{bmatrix} u_n(\mathbf{x}) \\ Q_n(\mathbf{x}) \\ P_{n\ell}(\mathbf{x}) \end{bmatrix} dS.$$

Again, we just show the result and omit the proof similar to that in Sect. 4.3.1.

Proposition 6 (Representation Theorem 4) *Consider the boundary-value problem (22), (23), and (25a) or (25b) for a bounded domain. The outward fluxes of mass, heat, and momentum through an arbitrary area A_g on ∂D_g are represented in terms of the corresponding Green function with respect to this area:*

$$\begin{aligned} \begin{bmatrix} \mathcal{M}(A_g) \\ \mathcal{Q}(A_g) \\ \mathcal{F}_\ell(A_g) \end{bmatrix} &= - \int_{\partial D_w} \begin{bmatrix} P_w u_n^{(P;A_g)} + \tau_w Q_n^{(P;A_g)} - u_{wj} P_{nj}^{(P;A_g)} \\ P_w u_n^{(T;A_g)} + \tau_w Q_n^{(T;A_g)} - u_{wj} P_{nj}^{(T;A_g)} \\ P_w u_n^{(-\ell;A_g)} + \tau_w Q_n^{(-\ell;A_g)} - u_{wj} P_{nj}^{(-\ell;A_g)} \end{bmatrix} dS \\ &- \int_{\partial D_g} \begin{bmatrix} \langle \zeta_n h^- G^{(P;A_g)} \rangle \\ \langle \zeta_n h^- G^{(T;A_g)} \rangle \\ \langle \zeta_n h^- G^{(-\ell;A_g)} \rangle \end{bmatrix} dS - \int_{A_g} \begin{bmatrix} \langle \zeta_n h \rangle \\ \langle \zeta_n (|\boldsymbol{\zeta}|^2 - \frac{5}{2}) h \rangle \\ \langle 2\zeta_n \zeta_j \ell_j h \rangle \end{bmatrix} dS, \end{aligned}$$

where ℓ is an arbitrary unit vector. The way of extension of h_{in} on $\partial D_g^{(i)}$ does not influence the relation (see footnote 15).

¹⁵The reason is slightly different from the case explained in footnote 12. The extended part of h^- is for $\zeta_n > 0$. By definition, $G^{(\alpha;\mathbf{x}_g)} = \delta(\mathbf{x} - \mathbf{x}_g)$, $(|\boldsymbol{\zeta}|^2 - \frac{5}{2})\delta(\mathbf{x} - \mathbf{x}_g)$, or $-2\zeta_j \ell_j \delta(\mathbf{x} - \mathbf{x}_g)$ ($\alpha = P, T$, or $-\ell$) in this range, so that the contributions of the extended part to the second and third terms on the right-hand side cancel out each other.

Table 3 Green functions for elemental sources at infinity for unbounded domain

Green function ^a	Corresponding elemental source	Note
$G^{(P;\infty)}(\mathbf{x}, \zeta)$	$g_w = 0 \quad h = 1$	—
$G^{(T;\infty)}(\mathbf{x}, \zeta)$	$g_w = 0 \quad h = (\zeta ^2 - \frac{5}{2})$	—
$G^{(\ell;\infty)}(\mathbf{x}, \zeta)$	$g_w = 0 \quad h = 2\zeta_i \ell_i$	$G^{(\ell;\infty)}(\mathbf{x}, \zeta) = -G^{(-\ell;\infty)}(\mathbf{x}, \zeta)$
$G^{(\underline{\Omega}(\ell);\infty)}(\mathbf{x}, \zeta)$	$g_w = 0 \quad h = 2\varepsilon_{ijk} \zeta_i \ell_j x_k$	$G^{(\underline{\Omega}(\ell);\infty)}(\mathbf{x}, \zeta) = -G^{(-\underline{\Omega}(\ell);\infty)}(\mathbf{x}, \zeta)$

^a ℓ is an arbitrary unit vector. $\underline{\Omega}(\ell) = (\Omega_{ij})$ is an alternating matrix defined by $\Omega_{ij} = -\varepsilon_{ijk} \ell_k$, where ε_{ijk} is Edington’s epsilon

4.3.3 Representation of Fluxes Through ∂D_g for an Unbounded Domain

We finally turn to the representation of the fluxes passing through ∂D_g for an unbounded domain and introduce the Green functions listed in Table 3. The main difference from the previous cases is the fact that they are the Green functions for the sources on the whole area of ∂D_g and that a new type of elemental source, which we shall call the rotation source around ℓ -axis, is introduced in the last line of the table. It should be reminded that in the present case h is required to satisfy (22) and that any linearized local Maxwellian satisfying (22) is limited to a linear combination of the elemental sources in Table 3.

Let us denote by $\rho_0 L^2 (2kT_0/m)^{1/2} \mathcal{M}(\infty)$, $\frac{1}{2} \rho_0 L^2 (2kT_0/m)^{3/2} \mathcal{Q}(\infty)$, $p_0 L^2 \mathcal{F}_\ell(\infty)$, and $p_0 L^3 \mathcal{T}_\ell(\infty)$ the mass, heat, momentum in the ℓ -direction, and angular momentum around the ℓ -axis transferred to a far field (or ∂D_g) per unit time. $p_0 L^2 \mathcal{F}_\ell(\infty)$ and $p_0 L^3 \mathcal{T}_\ell(\infty)$ may be regarded as the force and the torque (or the moment of force) acting on ∂D_g in the ℓ -direction. They are written in terms of u_n , Q_n and P_{nj} as

$$\begin{bmatrix} \mathcal{M}(\infty) \\ \mathcal{Q}(\infty) \\ \mathcal{F}_\ell(\infty) \\ \mathcal{T}_\ell(\infty) \end{bmatrix} = - \int_{\partial D_g} \begin{bmatrix} u_n(\mathbf{x}) \\ Q_n(\mathbf{x}) \\ P_{n\ell}(\mathbf{x}) \\ \ell_i \varepsilon_{ijk} x_j P_{nk}(\mathbf{x}) \end{bmatrix} dS.$$

Applying Proposition 2 to the pair of the solution of the boundary-value problem (22), (23), and (25) and the Green functions in Table 3 leads to the following:

Proposition 7 (Representation Theorem 5) Consider the boundary-value problem (22), (23), and (25c) for an unbounded domain. The outward fluxes of mass, heat, and linear and angular momentums through ∂D_g (i.e., the net fluxes toward the outer far field) are represented in terms of the corresponding Green function:

$$\begin{bmatrix} \mathcal{M}(\infty) \\ \mathcal{Q}(\infty) \\ \mathcal{F}_\ell(\infty) \\ \mathcal{T}_\ell(\infty) \end{bmatrix} = - \int_{\partial D_w} \begin{bmatrix} P_w u_n^{(P;\infty)} + \tau_w Q_n^{(P;\infty)} - u_{wj} P_{nj}^{(P;\infty)} \\ P_w u_n^{(T;\infty)} + \tau_w Q_n^{(T;\infty)} - u_{wj} P_{nj}^{(T;\infty)} \\ P_w u_n^{(-\ell;\infty)} + \tau_w Q_n^{(-\ell;\infty)} - u_{wj} P_{nj}^{(-\ell;\infty)} \\ P_w u_n^{(-\underline{\Omega}(\ell);\infty)} + \tau_w Q_n^{(-\underline{\Omega}(\ell);\infty)} - u_{wj} P_{nj}^{(-\underline{\Omega}(\ell);\infty)} \end{bmatrix} dS$$

$$- \int_{\partial D_g} \left\langle \zeta_n h^- \begin{bmatrix} G^{(P;\infty)} - 1 \\ G^{(T;\infty)} - (|\zeta|^2 - \frac{5}{2}) \\ G^{(-\ell;\infty)} + 2\zeta_j \ell_j \\ G^{(-\underline{\Omega}(\ell);\infty)} + 2\zeta_j \varepsilon_{jkl} \ell_k x_l \end{bmatrix} \right\rangle dS,$$

where ℓ is an arbitrary unit vector.

Practically, it is important to note that the above representation yields immediately that of the outward net fluxes through ∂D_w by changing the sign of the right-hand side, because of the conservation of mass, linear and angular momentums, and energy.

4.4 Reciprocity of the Fluxes Induced by the Green Functions

As a direct consequence of Propositions 4–7, we obtain a set of corollaries on the reciprocity of the Green functions, which we summarize here. Besides its significance by itself, the reciprocity will play a key role in our entropy theory to be developed in a separate paper. Before showing the corollaries, let us recall the notation convention immediately after (26). For instance, we denote by $\mathcal{M}^{(\alpha; B_w)}(A_w)$ and $\mathcal{Q}^{(\alpha; B_w)}(A_w)$ the dimensionless outward fluxes of mass and heat through area A_w induced by the Green function with respect to area B_w :

$$\begin{bmatrix} \mathcal{M}^{(\alpha; B_w)}(A_w) \\ \mathcal{Q}^{(\alpha; B_w)}(A_w) \end{bmatrix} = - \int_{A_w} \begin{bmatrix} u_n^{(\alpha; B_w)}(\mathbf{x}_0) \\ \mathcal{Q}_n^{(\alpha; B_w)}(\mathbf{x}_0) \end{bmatrix} dS_0 \quad (\alpha = P, T).$$

Throughout this subsection, ℓ and m are an arbitrary unit vector and the component in their direction ℓ will be indicated by subscripted ℓ and m , e.g., $\mathcal{F}_\ell = \mathcal{F}_j \ell_j$, $P_{nm} = P_{nj} m_j$.

Corollary 2 For any areas A_w, B_w on ∂D_w ,

$$\begin{bmatrix} \mathcal{M}^{(P; B_w)}(A_w) \\ \mathcal{Q}^{(P; B_w)}(A_w) \end{bmatrix} = \begin{bmatrix} \mathcal{M}^{(P; A_w)}(B_w) \\ \mathcal{Q}^{(P; A_w)}(B_w) \end{bmatrix}, \quad \begin{bmatrix} \mathcal{M}^{(T; B_w)}(A_w) \\ \mathcal{Q}^{(T; B_w)}(A_w) \end{bmatrix} = \begin{bmatrix} \mathcal{Q}^{(P; A_w)}(B_w) \\ \mathcal{Q}^{(T; A_w)}(B_w) \end{bmatrix}.$$

Proof Apply Proposition 4 to the boundary-value problem for $G^{(\alpha; B_w)}$ ($\alpha = P, T$). □

Corollary 3 For any $\mathbf{x}_0, \mathbf{x}_1 \in \partial D_g$ in a bounded domain,

$$\begin{aligned} & \begin{bmatrix} u_n^{(P; \mathbf{x}_0)}(\mathbf{x}_1), & u_n^{(T; \mathbf{x}_0)}(\mathbf{x}_1), & u_n^{(-m; \mathbf{x}_0)}(\mathbf{x}_1) \\ \mathcal{Q}_n^{(P; \mathbf{x}_0)}(\mathbf{x}_1), & \mathcal{Q}_n^{(T; \mathbf{x}_0)}(\mathbf{x}_1), & \mathcal{Q}_n^{(-m; \mathbf{x}_0)}(\mathbf{x}_1) \\ P_{n\ell}^{(P; \mathbf{x}_0)}(\mathbf{x}_1) - n_i \ell_i \delta(\mathbf{x}_1 - \mathbf{x}_0), & P_{n\ell}^{(T; \mathbf{x}_0)}(\mathbf{x}_1), & P_{n\ell}^{(-m; \mathbf{x}_0)}(\mathbf{x}_1) \end{bmatrix} \\ &= \begin{bmatrix} u_n^{(P; \mathbf{x}_1)}(\mathbf{x}_0), & \mathcal{Q}_n^{(P; \mathbf{x}_1)}(\mathbf{x}_0), & P_{nm}^{(P; \mathbf{x}_1)}(\mathbf{x}_0) - n_i m_i \delta(\mathbf{x}_0 - \mathbf{x}_1) \\ u_n^{(T; \mathbf{x}_1)}(\mathbf{x}_0), & \mathcal{Q}_n^{(T; \mathbf{x}_1)}(\mathbf{x}_0), & P_{nm}^{(T; \mathbf{x}_1)}(\mathbf{x}_0) \\ u_n^{(-\ell; \mathbf{x}_1)}(\mathbf{x}_0), & \mathcal{Q}_n^{(-\ell; \mathbf{x}_1)}(\mathbf{x}_0), & P_{nm}^{(-\ell; \mathbf{x}_1)}(\mathbf{x}_0) \end{bmatrix}. \end{aligned}$$

Corollary 4 For any $\mathbf{x}_w \in \partial D_w$ and $\mathbf{x}_g \in \partial D_g$ for a bounded domain, the following relation holds:

$$\begin{aligned} & \begin{bmatrix} u_n^{(P; \mathbf{x}_w)}(\mathbf{x}_g), & u_n^{(T; \mathbf{x}_w)}(\mathbf{x}_g), & u_n^{(-t; \mathbf{x}_w)}(\mathbf{x}_g) \\ \mathcal{Q}_n^{(P; \mathbf{x}_w)}(\mathbf{x}_g), & \mathcal{Q}_n^{(T; \mathbf{x}_w)}(\mathbf{x}_g), & \mathcal{Q}_n^{(-t; \mathbf{x}_w)}(\mathbf{x}_g) \\ P_{n\ell}^{(P; \mathbf{x}_w)}(\mathbf{x}_g), & P_{n\ell}^{(T; \mathbf{x}_w)}(\mathbf{x}_g), & P_{n\ell}^{(-t; \mathbf{x}_w)}(\mathbf{x}_g) \end{bmatrix} \\ &= \begin{bmatrix} u_n^{(P; \mathbf{x}_g)}(\mathbf{x}_w), & \mathcal{Q}_n^{(P; \mathbf{x}_g)}(\mathbf{x}_w), & P_{nt}^{(P; \mathbf{x}_g)}(\mathbf{x}_w) \\ u_n^{(T; \mathbf{x}_g)}(\mathbf{x}_w), & \mathcal{Q}_n^{(T; \mathbf{x}_g)}(\mathbf{x}_w), & P_{nt}^{(T; \mathbf{x}_g)}(\mathbf{x}_w) \\ u_n^{(-\ell; \mathbf{x}_g)}(\mathbf{x}_w), & \mathcal{Q}_n^{(-\ell; \mathbf{x}_g)}(\mathbf{x}_w), & P_{nt}^{(-\ell; \mathbf{x}_g)}(\mathbf{x}_w) \end{bmatrix}, \end{aligned}$$

where t is a unit vector tangential to ∂D_w at \mathbf{x}_w .

Proof Apply Proposition 5 to the boundary-value problem for $G^{(\alpha;x_0)}$ ($\alpha = P, T, -m$) by putting $x_g = x_1$ for Corollary 3 and to that for $G^{(\alpha;x_w)}$ ($\alpha = P, T, -t$) for Corollary 4. \square

Corollary 5 For any areas $A_g, B_g \subseteq \partial D_g$ in a bounded domain,

$$\begin{aligned} & \begin{bmatrix} \mathcal{M}^{(P;B_g)}(A_g), & \mathcal{M}^{(T;B_g)}(A_g), & \mathcal{M}^{(-m;B_g)}(A_g) \\ \mathcal{Q}^{(P;B_g)}(A_g), & \mathcal{Q}^{(T;B_g)}(A_g), & \mathcal{Q}^{(-m;B_g)}(A_g) \\ \mathcal{F}_\ell^{(P;B_g)}(A_g) + \ell_i \int_{A_g \cap B_g} n_i dS, & \mathcal{F}_\ell^{(T;B_g)}(A_g), & \mathcal{F}_\ell^{(-m;B_g)}(A_g) \end{bmatrix} \\ &= \begin{bmatrix} \mathcal{M}^{(P;A_g)}(B_g), & \mathcal{Q}^{(P;A_g)}(B_g), & \mathcal{F}_m^{(P;A_g)}(B_g) + m_i \int_{A_g \cap B_g} n_i dS \\ \mathcal{M}^{(T;A_g)}(B_g), & \mathcal{Q}^{(T;A_g)}(B_g), & \mathcal{F}_m^{(T;A_g)}(B_g) \\ \mathcal{M}^{(-t;A_g)}(B_g), & \mathcal{Q}^{(-t;A_g)}(B_g), & \mathcal{F}_m^{(-t;A_g)}(B_g) \end{bmatrix}. \end{aligned}$$

Corollary 6 For any areas $A_w \in \partial D_w$ and $A_g \in \partial D_g$ in a bounded domain,

$$\begin{bmatrix} \mathcal{M}^{(P;A_w)}(A_g), & \mathcal{M}^{(T;A_w)}(A_g) \\ \mathcal{Q}^{(P;A_w)}(A_g), & \mathcal{Q}^{(T;A_w)}(A_g) \\ \mathcal{F}_\ell^{(P;A_w)}(A_g), & \mathcal{F}_\ell^{(T;A_w)}(A_g) \end{bmatrix} = \begin{bmatrix} \mathcal{M}^{(P;A_g)}(A_w), & \mathcal{Q}^{(P;A_g)}(A_w) \\ \mathcal{M}^{(T;A_g)}(A_w), & \mathcal{Q}^{(T;A_g)}(A_w) \\ \mathcal{M}^{(-t;A_g)}(A_w), & \mathcal{Q}^{(-t;A_g)}(A_w) \end{bmatrix}.$$

Proof Apply Proposition 6 to the boundary-value problem for $G^{(\alpha;B_g)}$ ($\alpha = P, T, -m$) for Corollary 5 and to that for $G^{(\alpha;A_w)}$ ($\alpha = P, T$) for Corollary 6. \square

Corollary 7 The following reciprocal relation holds:

$$\begin{aligned} \mathcal{M}^{(T;\infty)}(\infty) &= \mathcal{Q}^{(P;\infty)}(\infty), & \mathcal{M}^{(-m;\infty)}(\infty) &= \mathcal{F}_m^{(P;\infty)}(\infty), \\ \mathcal{Q}^{(-m;\infty)}(\infty) &= \mathcal{F}_m^{(T;\infty)}(\infty), & \mathcal{F}_\ell^{(-m;\infty)}(\infty) &= \mathcal{F}_m^{(-t;\infty)}(\infty), \\ \mathcal{M}^{(-\underline{\Omega}(m);\infty)}(\infty) &= \mathcal{T}_m^{(P;\infty)}(\infty), & \mathcal{F}_\ell^{(-\underline{\Omega}(m);\infty)}(\infty) &= \mathcal{T}_m^{(-t;\infty)}(\infty), \\ \mathcal{Q}^{(-\underline{\Omega}(m);\infty)}(\infty) &= \mathcal{T}_m^{(T;\infty)}(\infty), & \mathcal{T}_\ell^{(-\underline{\Omega}(m);\infty)}(\infty) &= \mathcal{T}_m^{(-\underline{\Omega}(\ell);\infty)}(\infty). \end{aligned}$$

Corollary 8 For any $x \in \partial D_w$ and $A \subseteq \partial D_w$,

$$\begin{aligned} & \begin{bmatrix} \mathcal{M}^{(P;x)}(\infty), & \mathcal{M}^{(T;x)}(\infty), & \mathcal{M}^{(-m;x)}(\infty) \\ \mathcal{Q}^{(P;x)}(\infty), & \mathcal{Q}^{(T;x)}(\infty), & \mathcal{Q}^{(-m;x)}(\infty) \\ \mathcal{F}_\ell^{(P;x)}(\infty), & \mathcal{F}_\ell^{(T;x)}(\infty), & \mathcal{F}_\ell^{(-m;x)}(\infty) \\ \mathcal{T}_\ell^{(P;x)}(\infty), & \mathcal{T}_\ell^{(T;x)}(\infty), & \mathcal{T}_\ell^{(-m;x)}(\infty) \end{bmatrix} \\ &= - \begin{bmatrix} u_n^{(P;\infty)}(x), & Q_n^{(P;\infty)}(x), & P_{nm}^{(P;\infty)}(x) \\ u_n^{(T;\infty)}(x), & Q_n^{(T;\infty)}(x), & P_{nm}^{(T;\infty)}(x) \\ u_n^{(-t;\infty)}(x), & Q_n^{(-t;\infty)}(x), & P_{nm}^{(-t;\infty)}(x) \\ u_n^{(-\underline{\Omega}(\ell);\infty)}(x), & Q_n^{(-\underline{\Omega}(\ell);\infty)}(x), & P_{nm}^{(-\underline{\Omega}(\ell);\infty)}(x) \end{bmatrix}, \\ & \begin{bmatrix} \mathcal{M}^{(P;A)}(\infty), & \mathcal{M}^{(T;A)}(\infty) \\ \mathcal{Q}^{(P;A)}(\infty), & \mathcal{Q}^{(T;A)}(\infty) \\ \mathcal{F}_\ell^{(P;A)}(\infty), & \mathcal{F}_\ell^{(T;A)}(\infty) \\ \mathcal{T}_\ell^{(P;A)}(\infty), & \mathcal{T}_\ell^{(T;A)}(\infty) \end{bmatrix} = \begin{bmatrix} \mathcal{M}^{(P;\infty)}(A), & \mathcal{Q}^{(P;\infty)}(A) \\ \mathcal{M}^{(T;\infty)}(A), & \mathcal{Q}^{(T;\infty)}(A) \\ \mathcal{M}^{(-t;\infty)}(A), & \mathcal{Q}^{(-t;\infty)}(A) \\ \mathcal{M}^{(-\underline{\Omega}(\ell);\infty)}(A), & \mathcal{Q}^{(-\underline{\Omega}(\ell);\infty)}(A) \end{bmatrix}. \end{aligned}$$

Proof Apply Proposition 7 to the boundary-value problem for $G^{(\alpha;\infty)}$ ($\alpha = T, -\mathbf{m}, -\underline{\Omega}(\mathbf{m})$) for Corollary 7 and to that for $G^{(\alpha;x)}$ ($\alpha = P, T, -\mathbf{m}$) or $G^{(\beta:A)}$ ($\beta = P, T$) for Corollary 8. \square

5 Applications of the Representation Theorem

In this section, we present application examples of the representation theorem. Some of them (Examples 1, 2, and 5) have been discussed in the literature as the examples of the Onsager–Casimir reciprocity in the connection to the entropy production (e.g., [2, 8, 9, 42, 43]). It should be noted that they are recovered merely as byproducts of the representation theorem and that the discussion of the entropy production is entirely excluded from the present paper.

5.1 Mass and Heat Fluxes and the Force Acting on the Resting Bodies in an Unbounded Domain

Consider a group of N resting bodies (say, B_1, \dots, B_N) arranged in a finite region in the gas occupying an unbounded domain. In a far field, the gas is in the equilibrium state at rest with density ρ_0 and temperature T_0 . The respective bodies in the group may be a simple solid body or a condensed phase of the gas; they do not change in time their shape and surface temperature and have no surface velocity.

When the deviation from the reference equilibrium state is small, the problem can be linearized around the reference state. Then, as a direct application of Proposition 4 and Corollary 2 ($\partial D_w = \partial B_1 + \dots + \partial B_N, \partial D_g = \partial D_g^{(ii)}, A_w = \partial B_j, B_w = \partial B_k, u_{wi} = 0,$ and $h = 0$), we have the following statement for the mass and heat transferred to the resting bodies:

Corollary 9 (Mass and heat transferred to the resting bodies) *The mass and heat transferred to the body B_j ($j = 1, \dots, N$) per unit time can be expressed by the mass and heat flow distributions over the whole body surface ∂D_w induced by the Green functions for the pressure and temperature sources with respect to ∂B_j :*

$$\begin{bmatrix} \mathcal{M}(\partial B_j) \\ \mathcal{Q}(\partial B_j) \end{bmatrix} = - \int_{\partial D_w} \begin{bmatrix} u_n^{(P;\partial B_j)}(\mathbf{x}), & Q_n^{(P;\partial B_j)}(\mathbf{x}) \\ u_n^{(T;\partial B_j)}(\mathbf{x}), & Q_n^{(T;\partial B_j)}(\mathbf{x}) \end{bmatrix} \begin{bmatrix} P_w(\mathbf{x}) \\ \tau_w(\mathbf{x}) \end{bmatrix} dS.$$

In particular:

- (i) *if there is no condensed phase in the group of bodies, the mass transferred to each body vanishes and the above expression becomes much simpler as¹⁶*

$$\mathcal{Q}(\partial B_j) = - \int_{\partial D_w} Q_n^{(T;\partial B_j)}(\mathbf{x}) \tau_w(\mathbf{x}) dS. \tag{29}$$

¹⁶Consider a temperature field in the fluid dynamic limit, which is described by the Laplace equation. Let us denote by τ the perturbed temperature of the gas and by $\tau^{(\partial B_j)}$ the counterpart when the temperature source is put on ∂B_j , i.e., $\Delta \tau = 0, \Delta \tau^{(\partial B_j)} = 0, \tau = \tau_w$ on $\partial D_w, \tau^{(\partial B_j)} = 1$ on $\partial B_j,$ and $\tau^{(\partial B_j)} = 0$ on $\partial D_w \setminus \partial B_j.$ Then, the Green formula $\int_D (\tau \Delta \tau^{(\partial B_j)} - \tau^{(\partial B_j)} \Delta \tau) dx = \int_{\partial D} (\tau \nabla \tau^{(\partial B_j)} - \tau^{(\partial B_j)} \nabla \tau) \cdot \mathbf{n} dS$ is reduced to

$$0 = \int_{\partial D_w} \tau_w \nabla \tau^{(\partial B_j)} \cdot \mathbf{n} dS - \int_{\partial B_j} \nabla \tau \cdot \mathbf{n} dS + \int_{\partial D_g} (\tau \nabla \tau^{(\partial B_j)} - \tau^{(\partial B_j)} \nabla \tau) \cdot \mathbf{n} dS.$$

(ii) *the following reciprocity holds:*

$$\begin{bmatrix} \mathcal{M}^{(P;\partial B_k)}(\partial B_j), & \mathcal{M}^{(T;\partial B_k)}(\partial B_j) \\ \mathcal{Q}^{(P;\partial B_k)}(\partial B_j), & \mathcal{Q}^{(T;\partial B_k)}(\partial B_j) \end{bmatrix} = \begin{bmatrix} \mathcal{M}^{(P;\partial B_j)}(\partial B_k), & \mathcal{Q}^{(P;\partial B_j)}(\partial B_k) \\ \mathcal{M}^{(T;\partial B_j)}(\partial B_k), & \mathcal{Q}^{(T;\partial B_j)}(\partial B_k) \end{bmatrix}.$$

Example 1 (Evaporation from and condensation onto a volatile sphere) Consider the case where the group of the bodies is a single spherical condensed phase of the gas (for short, a volatile sphere). Denoting by S the surface of the volatile sphere, we have the expression

$$\begin{bmatrix} \mathcal{M}(S) \\ \mathcal{Q}(S) \end{bmatrix} = - \int_S \begin{bmatrix} u_n^{(P;S)}(\mathbf{x}), & Q_n^{(P;S)}(\mathbf{x}) \\ u_n^{(T;S)}(\mathbf{x}), & Q_n^{(T;S)}(\mathbf{x}) \end{bmatrix} \begin{bmatrix} P_w(\mathbf{x}) \\ \tau_w(\mathbf{x}) \end{bmatrix} dS$$

and the reciprocity

$$\mathcal{M}^{(T;S)}(S) = \mathcal{Q}^{(P;S)}(S).$$

Noted that $\mathcal{M}^{(P;S)}$ and $\mathcal{Q}^{(P;S)}$ are the dimensionless mass and heat passing through a point on S per unit time and area for $P_w = 1$ and $\tau_w = 0$, while $\mathcal{M}^{(T;S)}$ and $\mathcal{Q}^{(T;S)}$ are those for $P_w = 0$ and $\tau_w = 1$. Thus, as far as the mass and heat transfers are concerned, the general result is readily obtained by the study of two elemental situations, $P_w = 1, \tau_w = 0$ and $P_w = 0, \tau_w = 1$. The above reciprocity was numerically verified in the literature (e.g., [42, 43]).

We also have a statement on the force and torque on the group of resting bodies (not individual bodies) as a consequence of Proposition 7 and the conservation of linear and angular momentums:

Corollary 10 (Force and torque on the group of resting bodies) *The force and torque on the group of bodies can be obtained from the mass and heat flow distributions over the whole body surface ∂D_w induced by the Green functions for velocity and rotation sources in a far field. If we denote by $p_0 L^2 \mathcal{F}_\ell(\partial D_w)$ and $p_0 L^3 \mathcal{T}_\ell(\partial D_w)$ the force and torque on the group of bodies in the ℓ -direction. they are expressed as*

$$\begin{bmatrix} \mathcal{F}_\ell(\partial D_w) \\ \mathcal{T}_\ell(\partial D_w) \end{bmatrix} = - \int_{\partial D_w} \left(P_w(\mathbf{x}) \begin{bmatrix} u_n^{(\ell;\infty)}(\mathbf{x}) \\ u_n^{(\underline{\Omega}(\ell);\infty)}(\mathbf{x}) \end{bmatrix} + \tau_w(\mathbf{x}) \begin{bmatrix} Q_n^{(\ell;\infty)}(\mathbf{x}) \\ Q_n^{(\underline{\Omega}(\ell);\infty)}(\mathbf{x}) \end{bmatrix} \right) dS.$$

In particular, when there is no condensed phase in the group, i.e., when ∂D_w is a simple boundary, the mass transferred to each body vanishes and the force and torque on the group of bodies can be expressed only by the heat flow distribution:

$$\begin{bmatrix} \mathcal{F}_\ell(\partial D_w) \\ \mathcal{T}_\ell(\partial D_w) \end{bmatrix} = - \int_{\partial D_w} \tau_w(\mathbf{x}) \begin{bmatrix} Q_n^{(\ell;\infty)}(\mathbf{x}) \\ Q_n^{(\underline{\Omega}(\ell);\infty)}(\mathbf{x}) \end{bmatrix} dS.$$

Since τ and $\tau^{(\partial B_j)}$ decays with the rate of $|x|^{-1}$ as $|x| \rightarrow \infty$, the third integral vanishes, and we have

$$\int_{\partial D_w} \tau_w \nabla \tau^{(\partial B_j)} \cdot \mathbf{n} dS = \int_{\partial B_j} \nabla \tau \cdot \mathbf{n} dS.$$

The expression (29) is the generalization of this relation to the gas of arbitrary Knudsen number.

Proof By the linear and angular momentum conservations, $\mathcal{F}_\ell(\partial D_w) = -\mathcal{F}_\ell(\infty)$ and $\mathcal{T}_\ell(\partial D_w) = -\mathcal{T}_\ell(\infty)$. Use Proposition 7 for the representation of $\mathcal{F}_\ell(\infty)$ and $\mathcal{T}_\ell(\infty)$ by putting $u_{wi} = 0$ and $h = 0$. Finally use the parity of $G^{(\ell;\infty)}$ in Table 3. \square

Example 2 (Force acting on a simple solid body with non-uniform temperature [44]) Consider the case where the group of the bodies is a single simple solid body. Denoting by S the surface of the body, the dimensionless force acting on the body is given by

$$\mathcal{F}_\ell(S) = - \int_S \tau_w(\mathbf{x}) Q_n^{(\ell;\infty)}(\mathbf{x}) dS.$$

$Q_n^{(\ell;\infty)}$ is the dimensionless heat flow induced by $G^{(\ell;\infty)}$. This Green function is a solution of the problem of a uniform slow flow in the ℓ -direction past the body with the reference uniform upstream temperature T_0 (to be precise, the solution normalized by the upstream flow speed).

Extension of Corollaries 9 and 10 to the case of bodies with moving surface under the constraint of $\mathbf{u}_w \cdot \mathbf{n} = 0$ is straightforward. In the case of a bounded domain with the boundary composed only of the real boundary ($\partial D = \partial D_w$), Corollaries 9 and 10 hold as it is with a proper choice of the reference equilibrium state. On the other hand, if the gas is not necessarily in some resting equilibrium state in a far field, we need to use Proposition 4 or 7 directly. The next is such an example.

Example 3 (Thermophoresis) Consider an infinite expanse of a resting gas with the temperature distribution $T_0(1 + Cx_1)$ (C : a positive constant), in which a single simple solid body with temperature T_0 is located at the origin. In this situation, there occurs a force acting on the body, which is the present concern (see, e.g., [5–7] and the references therein). The sources on the boundary are given by $g_w = 0$ and $h = C[(|\boldsymbol{\zeta}|^2 - \frac{5}{2})x_1 - (\sqrt{\pi}/2)\text{Kn}\zeta_1 A(|\boldsymbol{\zeta}|)]$ in this situation, where A is the one already defined in Sect. 3.1. We denote by S the surface of the body and use Proposition 7 to obtain the relation

$$\mathcal{F}_\ell(S) = -\mathcal{F}_\ell(\infty) = -C \int_{\partial D_g} \left(x_1 Q_n^{(\ell;\infty)} + \frac{\sqrt{\pi}}{2} \text{Kn} \langle \zeta_n \zeta_1 A(|\boldsymbol{\zeta}|) G^{(\ell;\infty)} \rangle \right) dS.$$

Here the parity of $G^{(\ell;\infty)}$ in Table 3 and the conservation of momentum have been used.

In the meantime, since both h and $G^{(\ell;\infty)}$ solve (22), we can show in the same way as in the proof of Proposition 1 that $\int_{\partial D} \langle \zeta_n h^- G^{(\ell;\infty)} \rangle dS = 0$. By using this fact and $\partial D = \partial D_g \cup S$, we obtain an alternative expression¹⁷

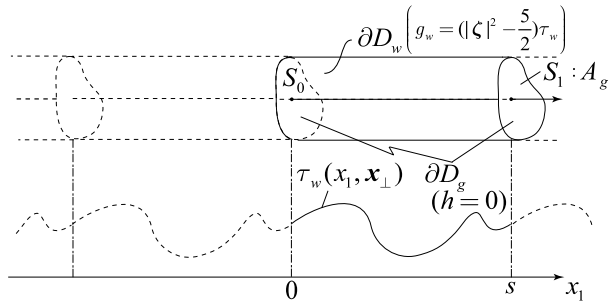
$$\mathcal{F}_\ell(S) = C \int_{\partial D_w} \left(x_1 Q_n^{(\ell;\infty)} + \frac{\sqrt{\pi}}{2} \text{Kn} \langle \zeta_n \zeta_1 A(|\boldsymbol{\zeta}|) G^{(\ell;\infty)} \rangle \right) dS.$$

5.2 Mass and Heat Fluxes Along Various Channels

Example 4 (Golse's theorem) Consider a gas in a straight pipe with a uniform cross-section. The temperature of the pipe is periodic and is constant in time. In the situation, a steady flow

¹⁷Our result is different from that in [8]. This is due to the fact that the term of $\zeta_1 A(|\boldsymbol{\zeta}|)$ is missing in the asymptotic form of the perturbed distribution function in that reference. The expression in [8] is incorrect.

Fig. 2 A straight pipe with a periodic temperature distribution



is induced in the gas by the non-uniformity of the temperature of the pipe. Golse proved that the induced flow field does not generate the mass flux through the pipe cross-section, by assuming that the temperature variation is so small that the linearization of the problem is allowed (Golse’s theorem [12]). We shall show below that we can reach the same conclusion with a slight extension of the statement by the application of Proposition 6.

Let us take the coordinate x_1 in the axial direction of the pipe as in Fig. 2. Let s be the period in this direction and D be the domain surrounded by the pipe wall and by the cross-sections S_0 and S_1 located respectively at $x_1 = 0$ and $x_1 = s$. Because $P_w = 0$, $\mathbf{u}_w = 0$, and $h = 0$, Proposition 6 yields, by putting $A_g = S_1$, the relation

$$\mathcal{M}(S_1) = - \int_{\partial D_w} \tau_w(x_1, \mathbf{x}_\perp) Q_n^{(P;S_1)}(x_1, \mathbf{x}_\perp) dS,$$

where $\mathbf{x}_\perp = (x_2, x_3)$, and here the Green function $G^{(P;S_1)}$ is the solution of the problem:

$$\zeta_i \frac{\partial G^{(P;S_1)}}{\partial x_i} = \frac{2}{\sqrt{\pi}} \frac{1}{\text{Kn}} \mathcal{L}(G^{(P;S_1)}), \tag{30a}$$

$$G^{(P;S_1)}(x_1, \mathbf{x}_\perp, \zeta) = \int_{\zeta_n^* < 0} \frac{|\zeta_n^*| E(\zeta^*)}{|\zeta_n| E(\zeta)} R_{\text{CR}}(\zeta^*, \zeta) G^{(P;S_1)}(x_1, \mathbf{x}_\perp, \zeta^*) d\zeta^*,$$

$$\zeta_n > 0, (x_1, \mathbf{x}_\perp) \in \partial D_w, \tag{30b}$$

$$G^{(P;S_1)}(s, \mathbf{x}_\perp, \zeta) = 1 + G^{(P;S_1)}(0, \mathbf{x}_\perp, \zeta) \quad \text{for } \zeta_1 < 0, \tag{30c}$$

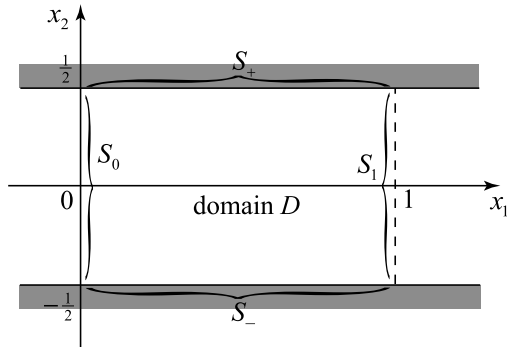
$$G^{(P;S_1)}(0, \mathbf{x}_\perp, \zeta) = G^{(P;S_1)}(s, \mathbf{x}_\perp, \zeta) - 1 \quad \text{for } \zeta_1 > 0, \tag{30d}$$

where R_{CR} is independent of \mathbf{x} . If the pipe wall is of the locally isotropic boundary [10, 12], we can seek the solution in the form¹⁸ $G^{(P;S_1)} = x_1/s + \zeta_1 \Phi(\mathbf{x}_\perp, \zeta)$, where Φ is even with respect to ζ_1 . Since Φ is even, $Q_n^{(P;S_1)} (= \zeta_n G^{(P;S_1)})$ vanishes, so that $\mathcal{M}(S_1) = 0$.

In the proof by Golse in the Appendix A.4 of [12], the Maxwell-type boundary condition with τ_w depending only on x_1 is considered. In contrast, we arrived at the same conclusion for a more general boundary condition and τ_w . In the sense, the present approach provides a slight extension of the applicable range of his statement.

¹⁸Here we assumed the similarity solution for brevity. Actually, however, this assumption is not necessary and we can show $Q_n^{(P;S_1)} = 0$ by the argument similar to that by Golse for the reduced problem (30).

Fig. 3 The channel between two parallel infinite plates and the domain D for the application of the representation theorem



Example 5 (Poiseuille, thermal transpiration, and Couette flows) Consider the steady behavior of the gas between two parallel plates located at $x_2 = \pm \frac{1}{2}$ in the following cases (see, e.g., [45–47]):

1. *Poiseuille flow* (PF, for short): two plates are at rest with a uniform constant temperature T_0 , and a uniform gradient of pressure in the x_1 -direction is imposed, i.e., the gas pressure is given by $p_0(1 + C_P x_1)$ with C_P being constant.
2. *Thermal transpiration* (TT, for short): two plates are at rest with a common temperature distribution $T_0(1 + C_T x_1)$ with C_T being constant.
3. *Couette flow* (CF, for short): two plates are at a uniform constant temperature T_0 . The upper plate ($x_2 = \frac{1}{2}$) is moving with the velocity $((2kT_0/m)^{1/2} u_{w1}, 0, 0)$, while the lower ($x_2 = -\frac{1}{2}$) is at rest.

We assume that $|C_P|$, $|C_T|$, and $|u_{w1}|$ are so small that the problems can be linearized around the resting equilibrium state with temperature T_0 and pressure p_0 . Further the state is assumed to be independent of x_3 . In what follows, we denote by putting the subscript PF, TT, and CF the solutions of the problems 1–3 normalized by the constants C_P , C_T , and u_{w1} .

Let D be the domain defined by $0 < x_1 < 1$, $-\frac{1}{2} < x_2 < \frac{1}{2}$, and $0 < x_3 < 1$ (see Fig. 3). Let S_0 and S_1 the cross-section (per unit length in the x_3 -direction) of the channel at $x_1 = 0$ and $x_1 = 1$. Let S_{\pm} be the surface of the plates at $x_2 = \pm \frac{1}{2}$ in the range of $0 < x_1 < 1$ and $0 < x_3 < 1$. It is easy to check that the solution of problem TT solves the problem in D with $g_w = x_1(|\zeta|^2 - \frac{5}{2})$, $h(x_1 = 0) = 0$, and $h(x_1 = 1) = |\zeta|^2 - \frac{5}{2}$, while that of problem CF solves the problem in D with $g_w = 2\zeta_1 \delta(x_2 - \frac{1}{2})$ and $h = 0$. With these in mind, we apply Proposition 6 by putting $A_g = S_1$ to obtain

$$\mathcal{M}_{TT}(S_1) = \pm \int_{S_{\pm}} x_1 Q_2^{(P;S_1)} dS + Q^{(P;S_1)}(S_1), \quad \mathcal{M}_{CF}(S_1) = -\mathcal{F}_1^{(P;S_1)}(S_+).$$

As in Example 4, we assume the plates are of the locally isotropic boundary, so that the Green function $G^{(P;S_1)}$ is given in the form of $G^{(P;S_1)} = x_1 + \zeta_1 \Phi(x_2, \zeta_2, |\zeta|)$. Since Φ is even in ζ_1 , $Q_2^{(P;S_1)}$ in the first equality vanishes. Finally noting that the normalized solution of the Poiseuille flow problem is the Green function $G^{(P;S_1)}$, we can rewrite the above relations as

$$\mathcal{M}_{TT}(S_1) = \mathcal{Q}_{PF}(S_1), \quad \mathcal{M}_{CF}(S_1) = -\mathcal{F}_{1PF}(S_+),$$

which show the cross relations among the three basic flows (see, e.g., [2]).

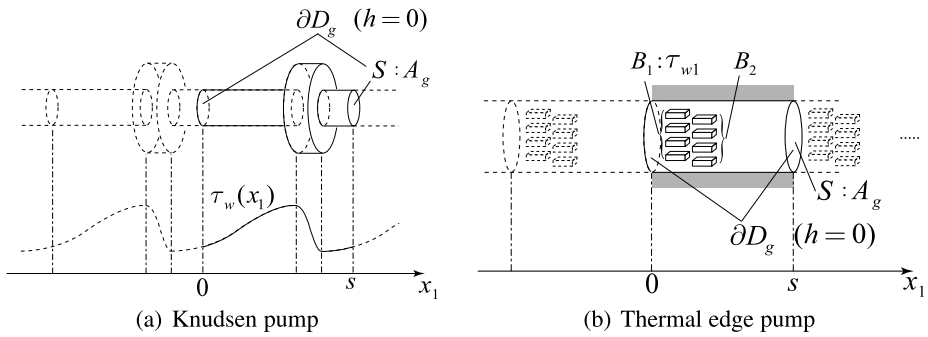


Fig. 4 Schematics of thermal pumps

Example 6 (Thermal pumps) Consider a gas in the pipe with periodic ditches as in Fig. 4(a). The temperature distribution of the pipe surface is constant in time and is periodic in x_1 with the same period s as the geometric structure. It is known that a one-way flow is induced by the non-uniform temperature in the pipe [48], which is the driving mechanism of the so-called *Knudsen pump* [10, 49]. When the perturbed temperature of the pipe surface is small enough, we can apply Proposition 6 to the gas domain D (the unit stage of the pump drawn in solid lines in the figure) to have the relation

$$\mathcal{M}(S) = - \int_{\partial D_w} \tau_w(\mathbf{x}) Q_n^{(P;S)}(\mathbf{x}) dS,$$

because $P_w = 0$, $\mathbf{u}_w = 0$, and $h = 0$, where S is the cross-section at $x_1 = s$. Thus the mass flux by the one-way flow induced in the pump can be expressed by the heat flow distribution on the pipe surface of the Green function $G^{(P;S)}$ for the pressure source on S .

Even when the pipe is a simple straight one with a uniform temperature distribution, a one-way flow can be induced if two arrays of uniformly heated and unheated plates, say B_1 and B_2 , are put inside the channel periodically with respect to x_1 with period s (see Fig. 4(b)). The pipe equipped with such arrays of plates is called the *thermal edge pump* [10, 50]. For simplicity, let the temperature of the array B_2 be the same as the pipe temperature T_0 . Then, if the uniform perturbed temperature τ_{w1} of the array B_1 is small enough, we can apply Proposition 6 to the gas domain D (the unit stage of the pump drawn in solid lines in the figure) to have the relation

$$\mathcal{M}(S) = \tau_{w1} Q^{(P;S)}(\partial B_1),$$

because $\tau_w(\mathbf{x}) = \tau_{w1}$ on ∂B_1 and $\tau_w = 0$ both on ∂B_2 and on the pipe surface, $P_w = 0$, $\mathbf{u}_w = 0$, and $h = 0$, where S is the cross-section at $x_1 = s$. Thus the mass flux by the one-way flow induced in the pump can be obtained by knowing the heat that the array B_1 receives in the flow of the Green function $G^{(P;S)}$ for the pressure source on S .

6 Conclusion

In the present paper, we first established a symmetric relation (14) that holds widely between two steady problems of the linearized Boltzmann equation in Sect. 2. We also presented a concrete set of situations where the required condition is seen to be fulfilled by a rather

simple argument: the bounded domain, one-dimensional half-space, and three dimensional unbounded domain with a finite confinement of bodies. Then, in Sect. 3, we showed some application examples to the second situation (one-dimensional half-space problems). In particular, four unknown relations among the Knudsen layer problems were newly obtained.

A remarkable difference of the first and third situations from the second was the fact that we may deal with the source on the boundary separately in the former. Making use of this property, we developed in Sect. 4 a unified approach to the first and third situations, on the basis of the symmetric relation, to establish general expressions of the mass, linear and angular momentum, and heat fluxes, which we called the representation theorem. The theorem tells that the problem of finding a flux on the boundary is reduced to finding the solution of a single elemental problem in the considered domain, which we called the *Green function*, the response of the gas system to a proper elemental disturbance from the surroundings. These disturbances are the pressure, velocity, rotation, or temperature source put *on the boundary*. We finally presented some application examples of the theorem in Sect. 5.

Some of the presented examples show the recovery of the cross relations that have been discussed in the literature as the Onsager–Casimir reciprocity on the basis of entropy production. It suggests some relation of the present approach to that based on the entropy production (e.g. [9, 19, 38, 51]). We shall discuss the issue in a separate paper, where the corollaries in Sect. 4.4 will play a key role. Here we merely stress that the present straightforward approach based on (14) is widely applicable and yields useful relations or expressions without any connection to the entropy production argument.

Appendix A: Reflection Kernel R in the Linearized Problem

We summarize the properties of the reflection kernel R (R_{CR} and R_{PR}) for the linearized problem (see Appendix A.9 in [10]).

Properties of R_{CR}

1. $R_{CR}(\zeta^*, \zeta; \mathbf{x}) \geq 0$ for $\zeta_n^* < 0, \zeta_n > 0$.
2. $\int_{\zeta_n > 0} R_{CR}(\zeta^*, \zeta; \mathbf{x}) d\zeta = 1$ for $\zeta_n^* < 0$.
3. Let φ be $\varphi = c_0 + c_i \zeta_i + c_4 |\zeta|^2$, where c_0, c_i , and c_4 are independent of ζ . Among such φ , only $\varphi = c_0$ satisfies the relation

$$\varphi(\mathbf{x}, \zeta) E(\zeta) = \int_{\zeta_n^* < 0} \frac{|\zeta_n^*|}{|\zeta_n|} R_{CR}(\zeta^*, \zeta; \mathbf{x}) \varphi(\mathbf{x}, \zeta^*) E(\zeta^*) d\zeta^* \quad \text{for } \zeta_n > 0.$$

Note that the second property corresponds to the condition $\langle \zeta_n \phi \rangle = 0$ of no flow across the boundary. The third property corresponds to the natural requirement that in a resting container with a uniform temperature the resting equilibrium state with the same temperature is established. The specular reflection is excluded from R_{CR} by this property.

Properties of R_{PR}

1. $R_{PR}(\zeta^*, \zeta; \mathbf{x}) \geq 0$ for $\zeta_n^* < 0, \zeta_n > 0$.
2. For a certain given function $g_0(\mathbf{x}, \zeta) \geq 0$ defined for $\zeta_n > 0$,

$$E(\zeta) = g_0(\mathbf{x}, \zeta) + \int_{\zeta_n^* < 0} \frac{|\zeta_n^*|}{|\zeta_n|} R_{PR}(\zeta^*, \zeta; \mathbf{x}) E(\zeta^*) d\zeta^* \quad \text{for } \zeta_n > 0.$$

3. Let φ be $\varphi = c_0 + c_i \zeta_i + c_4 |\boldsymbol{\zeta}|^2$, where c_0, c_i , and c_4 are independent of $\boldsymbol{\zeta}$. Among such φ , only $\varphi = 0$ satisfies the relation

$$\varphi(\mathbf{x}, \boldsymbol{\zeta}) E(\boldsymbol{\zeta}) = \int_{\zeta_n^* < 0} \frac{|\zeta_n^*|}{|\zeta_n|} R_{PR}(\boldsymbol{\zeta}^*, \boldsymbol{\zeta}; \mathbf{x}) \varphi(\mathbf{x}, \boldsymbol{\zeta}^*) E(\boldsymbol{\zeta}^*) d\boldsymbol{\zeta}^* \quad \text{for } \zeta_n > 0.$$

The second and third properties are the counterpart to the third of R_{CR} . They are a natural requirement that in a resting volatile container with a uniform temperature the resting equilibrium state with the same temperature and the corresponding saturation gas pressure is established.

Appendix B: Condition of Detailed Balance

Besides the fundamental properties in Appendix A, the kernel R is often required to satisfy the so-called *condition of detailed balance*. For instance, the most widely used conditions, such as the diffuse reflection, Maxwell-type, and Cercignani–Lampis conditions, are known to satisfy that condition. [13]

The condition of detailed balance for the kernel R in the linear problem reads

$$|\zeta_n^*| R(\boldsymbol{\zeta}^*, \boldsymbol{\zeta}; \mathbf{x}) E(\boldsymbol{\zeta}^*) = |\zeta_n| R(-\boldsymbol{\zeta}, -\boldsymbol{\zeta}^*; \mathbf{x}) E(\boldsymbol{\zeta}) \quad \text{for } \zeta_n > 0, \zeta_n^* < 0. \tag{31}$$

With this property, we can prove the following:

Lemma 4 *Let ϕ^A and ϕ^B obey the condition (4) with $g_w = g_w^A$ and $g_w = g_w^B$ respectively. If the kernel R satisfies the condition of detailed balance (31), the following equality holds:*

$$\langle \zeta_n (\phi^{B-} - g_w^{B-}) (\phi^A - g_w^A) \rangle = 0.$$

Proof We denote $\phi^\alpha - g_w^\alpha$ by ψ^α ($\alpha = A, B$) for brevity and rewrite the left-hand side as

$$\langle \zeta_n \psi^{B-} \psi^A \rangle = \int_{\zeta_n > 0} \zeta_n \psi^{B-} \psi^A E(\boldsymbol{\zeta}) d\boldsymbol{\zeta} - \int_{\zeta_n > 0} \zeta_n \psi^B \psi^{A-} E(\boldsymbol{\zeta}) d\boldsymbol{\zeta}. \tag{32}$$

The first term is rewritten by the use of (4) for ψ^A as

$$\int_{\zeta_n > 0} \zeta_n \psi^{B-} \psi^A E(\boldsymbol{\zeta}) d\boldsymbol{\zeta} = \int_{\zeta_n > 0} \left[\int_{\zeta_n^* < 0} |\zeta_n^*| R(\boldsymbol{\zeta}^*, \boldsymbol{\zeta}; \mathbf{x}) E(\boldsymbol{\zeta}^*) \psi^{A*} d\boldsymbol{\zeta}^* \right] \psi^{B-} d\boldsymbol{\zeta},$$

while the second term is rewritten as

$$\begin{aligned} \int_{\zeta_n > 0} \zeta_n \psi^B \psi^{A-} E(\boldsymbol{\zeta}) d\boldsymbol{\zeta} &= \int_{\zeta_n > 0} \left[\int_{\zeta_n^* < 0} |\zeta_n^*| R(\boldsymbol{\zeta}^*, \boldsymbol{\zeta}; \mathbf{x}) E(\boldsymbol{\zeta}^*) \psi^{B*} d\boldsymbol{\zeta}^* \right] \psi^{A-} d\boldsymbol{\zeta} \\ &= \int_{\zeta_n^* < 0} \left[\int_{\zeta_n > 0} |\zeta_n| R(-\boldsymbol{\zeta}, -\boldsymbol{\zeta}^*; \mathbf{x}) E(\boldsymbol{\zeta}) \psi^{B-} d\boldsymbol{\zeta} \right] \psi^{A*} d\boldsymbol{\zeta}^*. \end{aligned}$$

Thus, the first and second terms cancel out each other because of the detailed balance (31). \square

Acknowledgements This work is supported in part by KAKENHI (No. 19560066) from JSPS and by the research grant from the UBE Foundation. This work was motivated by the interest in the Onsager reciprocity that the author shared with Professors Yoshio Sone and Tai-Ping Liu. The author expresses his sincere thanks to Professor Sone for his encouragement, helpful discussions, and comments on the manuscript. This work was finished during the stay at the Institute of Mathematics, Academia Sinica. The author expresses his sincere thanks to Professor Liu for the hospitality and his continuing interest in this work.

References

1. Kennard, E.H.: *Kinetic Theory of Gases*. McGraw-Hill, New York (1938)
2. Loyalka, S.K.: Kinetic theory of thermal transpiration and mechanocaloric effect I. *J. Chem. Phys.* **55**, 4497–4503 (1971)
3. Bakanov, S.P., Vysotskij, V.V., Deryaguin, B.V., Roldughin, V.I.: Thermal polarization of bodies in the rarefied flow. *J. Nonequilib. Thermodyn.* **8**, 75–83 (1983)
4. Takata, S., Sone, Y., Aoki, K.: Numerical analysis of a uniform flow of a rarefied gas past a sphere on the basis of the Boltzmann equation for hard-sphere molecules. *Phys. Fluids A* **5**, 716–737 (1993)
5. Epstein, P.S.: Zur Theorie des Radiometer. *Z. Phys.* **54**, 537–563 (1929)
6. Sone, Y.: Flow induced by thermal stress in rarefied gas. *Phys. Fluids* **15**, 1418–1423 (1972)
7. Takata, S., Aoki, K., Sone, Y.: Thermophoresis of a sphere with a uniform temperature: numerical analysis of the Boltzmann equation for hard-sphere molecules. In: Shizgal, B.D., Weaver, D.P. (eds.) *Rarefied Gas Dynamics*, pp. 626–639. AIAA, Washington (1994)
8. Sharipov, F.: Onsager–Casimir reciprocity relations for open gaseous systems at arbitrary rarefaction II. Application of the theory for single gas. *Physica A* **203**, 457–485 (1994)
9. Zhdanov, V.M., Roldughin, V.I.: Non-equilibrium thermodynamics and kinetic theory of rarefied gases. *Phys. Uspekhi* **41**, 349–378 (1998)
10. Sone, Y.: *Molecular Gas Dynamics*. Birkhäuser, Boston (2007). Supplementary Notes and Errata: Kyoto University Research Information Repository (<http://hdl.handle.net/2433/66098>)
11. Sone, Y.: Asymptotic theory of flow of rarefied gas over a smooth boundary I. In: Trilling, L., Wachman, H.Y. (eds.) *Rarefied Gas Dynamics*, vol. 1, pp. 243–253. Academic Press, New York (1969)
12. Sone, Y.: *Kinetic Theory and Fluid Dynamics*. Birkhäuser, Boston (2002). Supplementary Notes and Errata: Kyoto University Research Information Repository (<http://hdl.handle.net/2433/66099>)
13. Cercignani, C., *The Boltzmann Equation and Its Application*, Sect. III.3. Springer, New York (1988)
14. Liu, T.-P., Yu, S.-H.: The Green’s function and large-time behavior of solutions for the one-dimensional Boltzmann equation. *Commun. Pure Appl. Math.* **57**, 1543–1608 (2004)
15. Sunakawa, S.: *Theoretical Electromagnetism*. Kinokuniya, Tokyo (1999) (in Japanese)
16. Jeans, J.: *The Mathematical Theory of Electricity and Magnetism*, 5th edn. reprinted. Cambridge University Press, New York (1960)
17. Panofsky, W.K.H., Phillips, M.: *Classical Electricity and Magnetism*, 2nd edn. reprinted. Addison-Wesley, Tokyo (1962)
18. Chandrasekhar, S.: *Radiative Transfer*. Dover, New York (1960)
19. Sharipov, F.: Onsager–Casimir reciprocity relations for open gaseous systems at arbitrary rarefaction I. General theory for single gas. *Physica A* **203**, 437–456 (1994)
20. Sharipov, F.: Onsager–Casimir reciprocal relations based on the Boltzmann equation and gas-surface interaction: single gas. *Phys. Rev. E* **73**, 026110 (2006)
21. McCourt, F.R.W., Beenakker, J.J.M., Köhler, W.E., Kušćer, I.: *Nonequilibrium Phenomena in Polyatomic Gases*, vol. 2, Chap. 16. Clarendon Press, Oxford (1991)
22. Grad, H.: Singular and nonuniform limits of solutions of the Boltzmann equation. In: Bellman, R., Birkhoff, G., Abu-Shumays, I. (eds.) *Transport Theory*, pp. 269–308. AMS, Providence (1969)
23. Imai, I.: Some applications of function theory to fluid dynamics. In: 2nd International JSME Symposium on Fluid Machinery and Fluidics, Tokyo, pp. 15–23 (1972)
24. Tatsumi, T.: *Fluid Mechanics*. Baifukan, Tokyo (1982) (in Japanese)
25. Happel, J., Brenner, H.: *Low Reynolds Number Hydrodynamics*, 2nd edn. Noordhoff International, Leyden (1973)
26. Sobolev, S.L.: *Partial Differential Equations of Mathematical Physics*. Dover, New York (1989)
27. Sone, Y.: Thermal creep in rarefied gas. *J. Phys. Soc. Jpn.* **21**, 1836–1837 (1966)
28. Ohwada, T., Sone, Y., Aoki, K.: Numerical analysis of the shear and thermal creep flows of a rarefied gas over a plane wall on the basis of the linearized Boltzmann equation for hard-sphere molecules. *Phys. Fluids A* **1**, 1588–1599 (1989)

29. Siewert, C.E.: Viscous-slip, thermal-slip, and temperature jump coefficients as defined by the linearized Boltzmann equation and the Cercignani–Lampis boundary condition. *Phys. Fluids* **15**, 1696–1701 (2003)
30. Maslova, N.B.: Kramers problem in the kinetic theory of gases. *USSR Comput. Math. Phys.* **22**, 208–219 (1982)
31. Bardos, C., Caflisch, R.E., Nicolaenko, B.: The Milne and Kramers problems for the Boltzmann equation of a hard sphere gas. *Commun. Pure Appl. Math.* **39**, 323–352 (1986)
32. Coron, F., Golse, F., Sulem, C.: A classification of well-posed kinetic layer problems. *Commun. Pure Appl. Math.* **41**, 409–435 (1988)
33. Golse, F., Poupaud, F.: Stationary solutions of the linearized Boltzmann equation in a half-space. *Math. Methods Appl. Sci.* **11**, 483–502 (1989)
34. Sone, Y., Onishi, Y.: Kinetic theory of evaporation and condensation—hydrodynamic equation and slip boundary condition. *J. Phys. Soc. Jpn.* **44**, 1981–1994 (1978)
35. Yasuda, S., Takata, S., Aoki, K.: Numerical analysis of the shear flow of a binary mixture of hard-sphere gases over a plane wall. *Phys. Fluids* **16**, 1989–2003 (2004)
36. Pao, Y.P.: Application of kinetic theory to the problem of evaporation and condensation. *Phys. Fluids* **14**, 306–312 (1971)
37. Sone, Y., Ohwada, T., Aoki, K.: Evaporation and condensation on a plane condensed phase: numerical analysis of the linearized Boltzmann equation for hard-sphere molecules. *Phys. Fluids A* **1**, 1398–1405 (1989)
38. Bedeaux, D., Hermans, L.J.F., Ytrehus, T.: Slow evaporation and condensation. *Physica A* **169**, 263–280 (1990)
39. Sone, Y., Ohwada, T., Aoki, K.: Temperature jump and Knudsen layer in a rarefied gas over a plane wall: numerical analysis of the linearized Boltzmann equation for hard-sphere molecules. *Phys. Fluids A* **1**, 363–370 (1989)
40. Yasuda, S., Takata, S., Aoki, K.: Evaporation and condensation of a binary mixture of vapors on a plane condensed phase: numerical analysis of the linearized Boltzmann equation. *Phys. Fluids* **17**, 047105 (2005)
41. Takata, S., Yasuda, S., Aoki, K., Kosuge, S.: Temperature, pressure, and concentration jumps for a binary mixture of vapors on a plane condensed phase: numerical analysis of the linearized Boltzmann equation. *Phys. Fluids* **18**, 067102 (2006)
42. Chernyak, V.G., Margilevskiy, A.Ye.: The kinetic theory of heat and mass transfer from a spherical particle in a rarefied gas. *Int. J. Heat Mass Transfer* **32**, 2127–2134 (1989)
43. Takata, S., Sone, Y., Lhuillier, D., Wakabayashi, M.: Evaporation from or condensation onto a sphere: numerical analysis of the Boltzmann equation for hard-sphere molecules. *Comput. Math. Appl.* **35**, 193–214 (1998)
44. Takata, S., Sone, Y.: Flow induced around a sphere with a non-uniform surface temperature in a rarefied gas, with application to the drag and thermal force problems of a spherical particle with an arbitrary thermal conductivity. *Eur. J. Mech. B Fluids* **14**, 487–518 (1995)
45. Ohwada, T., Sone, Y., Aoki, K.: Numerical analysis of the Poiseuille and thermal transpiration flows between two parallel plates on the basis of the Boltzmann equation for hard-sphere molecules. *Phys. Fluids A* **1**, 2042–2049 (1989)
46. Sone, Y., Takata, S., Ohwada, T.: Numerical analysis of the plane Couette flow of a rarefied gas on the basis of the linearized Boltzmann equation for hard-sphere molecules. *Eur. J. Mech. B Fluids* **9**, 273–288 (1990)
47. Garcia, R.D.M., Siewert, C.E.: The linearized Boltzmann equation with Cercignani–Lampis boundary conditions: basic flow problems in a plane channel. *Eur. J. Mech. B Fluids* **28**, 387–396 (2009)
48. Sone, Y., Waniguchi, Y., Aoki, K.: One-way flow of a rarefied gas induced in a channel with a periodic temperature distribution. *Phys. Fluids* **8**, 2227–2235 (1996)
49. Knudsen, M.: Thermischer Molekulardruck der Gase in Röhren. *Ann. Phys. (Leipzig)* **33**, 1435–1448 (1910)
50. Sugimoto, H., Sone, Y.: Vacuum pump without a moving part driven by thermal edge flow. In: Capitelli, M. (ed.) *Rarefied Gas Dynamics*, pp. 168–173. AIP, Melville (2005)
51. Bedeaux, D.: Nonequilibrium thermodynamics and statistical physics of surfaces. In: Prigogine, I., Rice, S.A. (eds.) *Advances in Chemical Physics*, vol. LXIV, pp. 47–109. Wiley, New York (1986)